

HEAT AND MASS TRANSFER MODELING OF VACUUM INSULATED VESSEL STORING CRYOGENIC LIQUID IN LOSS OF VACUUM ACCIDENT

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CONTENTS

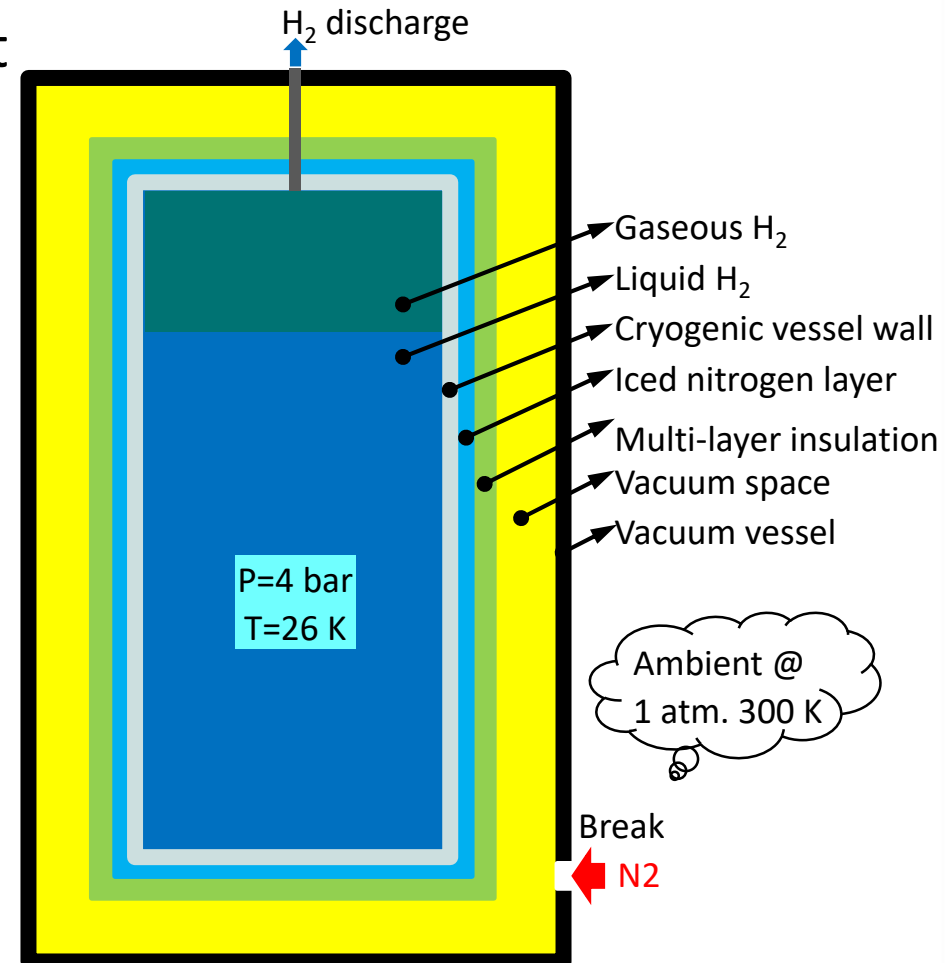
- Background
- Basic assumptions of vacuum insulated vessel in LOVA
- Material properties at cryogenic temperatures
- Heat and mass transfer models
 - Lumped parameter model
 - Dynamic venting flow at break
 - Mass transfer model
 - Convection model
 - Radiation model
 - Pool boiling heat transfer model
 - Conduction models
- Governing equation group (ODEs)
- Calculation results
- Conclusions

Background

- ❑ Cryogenics technology ($T < 120$ K) in applications of superconductor, aerospace, hydrogen technology etc.
- ❑ Application examples,
 - ❑ LHe of LHC of CERN
 - ❑ massive cryogenic storage system of NASA
 - ❑ shaft seal cooling system of large-scale device
 - ❑ liquid hydrogen application
- ❑ Standing and mobile hydrogen systems
 - ❑ LH2 storage system
 - ❑ H2 refueling station
 - ❑ LH2 powered LKWs, agriculture machinery, ships, airplanes etc.

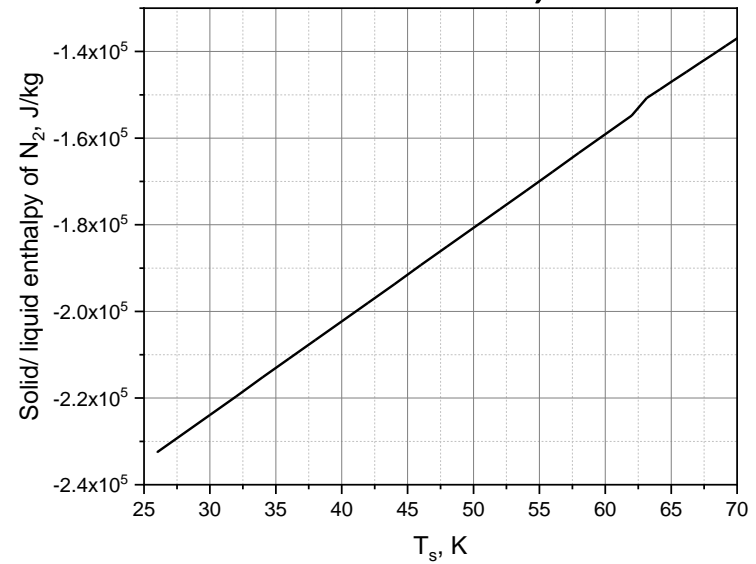
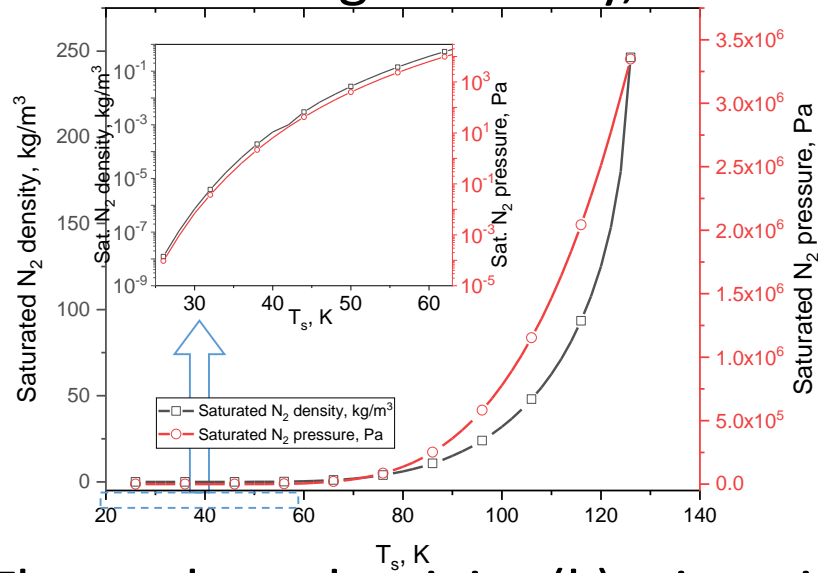
Basic assumptions of vacuum insulated vessel in LOVA

- ❑ What is LOVA: **l**oss of **v**acuum **a**ccident
- ❑ LOVA scenario: 10^{-5} mbar to 1 atm.
- ❑ Venting gas deposition occurs at the cryogenic temperature of outer wall surface of cryogenic vessel
- ❑ Iced layer can be on cryogenic vessel wall, or, in inner layers of MLI
- ❑ Venting gas is N_2 at 300 K and 1 atm.
- ❑ Safety discharge of H_2 activated at 4 bar
- ❑ Temperature of saturated two phase H_2 stays at 26 K uniformly

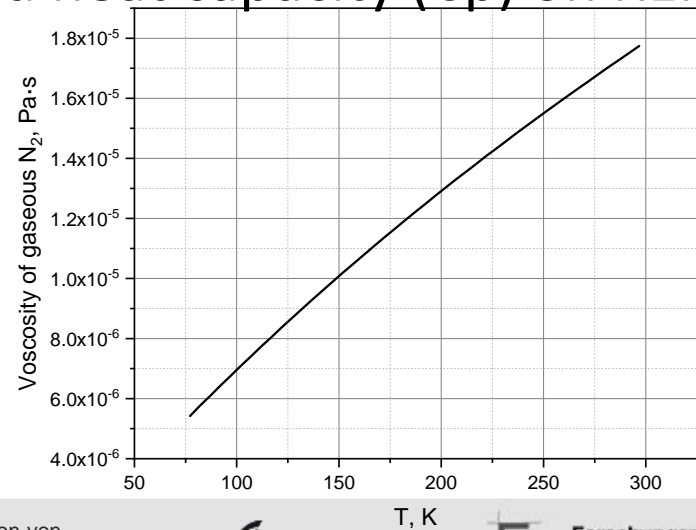
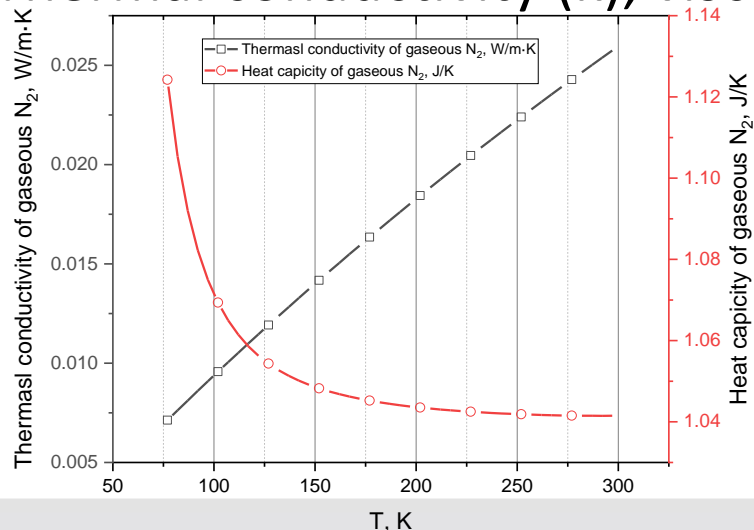


Material properties – gaseous N₂

Saturated N₂ gas density, enthalpy of iced N₂ or condensate, on REFPROP

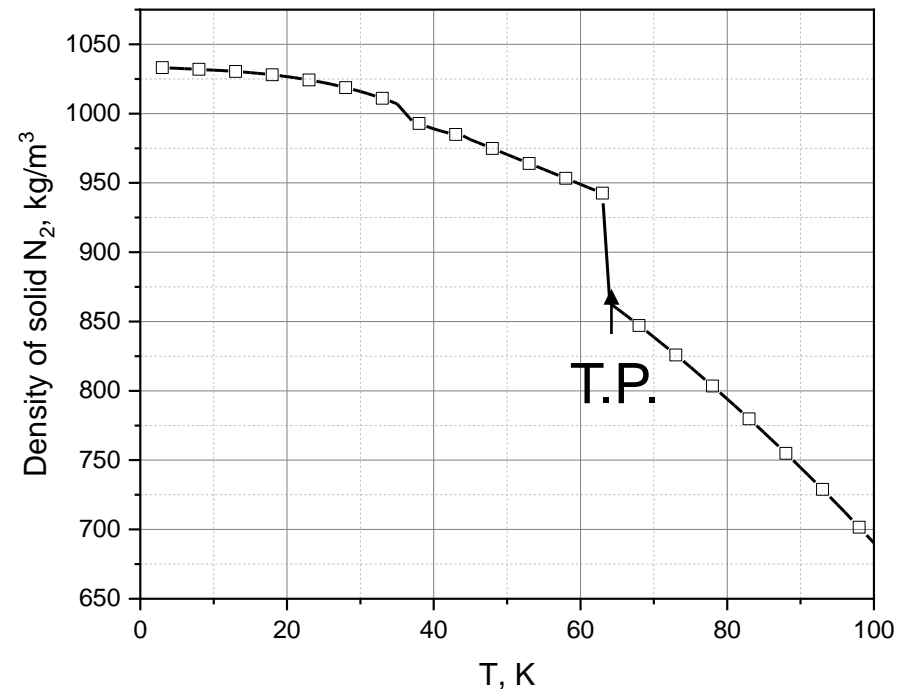
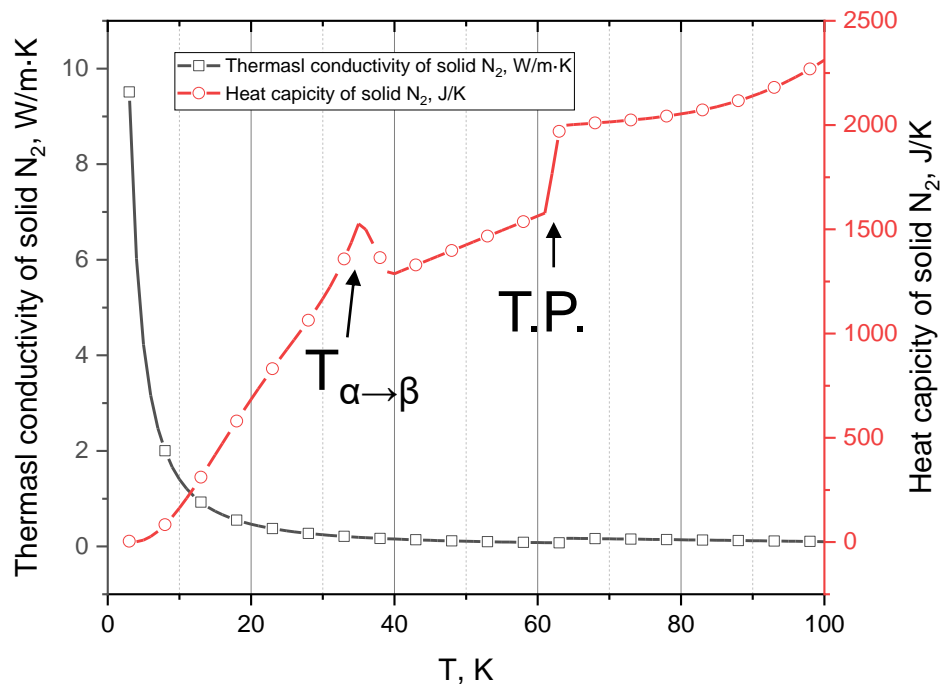


Thermal conductivity (k), viscosity (μ) and heat capacity (Cp) on REFPROP



Material properties – solid N₂

- Temperature dependent thermal conductivity k_{sn} , heat capacity Cp_{sn} and density ρ_{sn} – piecewise fitted



- *Solid and liquid nitrogen*, T.A. Scott, *Physics Reports* 27, 89 – 157 (1976), ISSN 0370-1573

- *NIST REFPROP*

Material properties –MLI

❑ What is MLI: **m**ulti-**l**ayer **i**nsulator

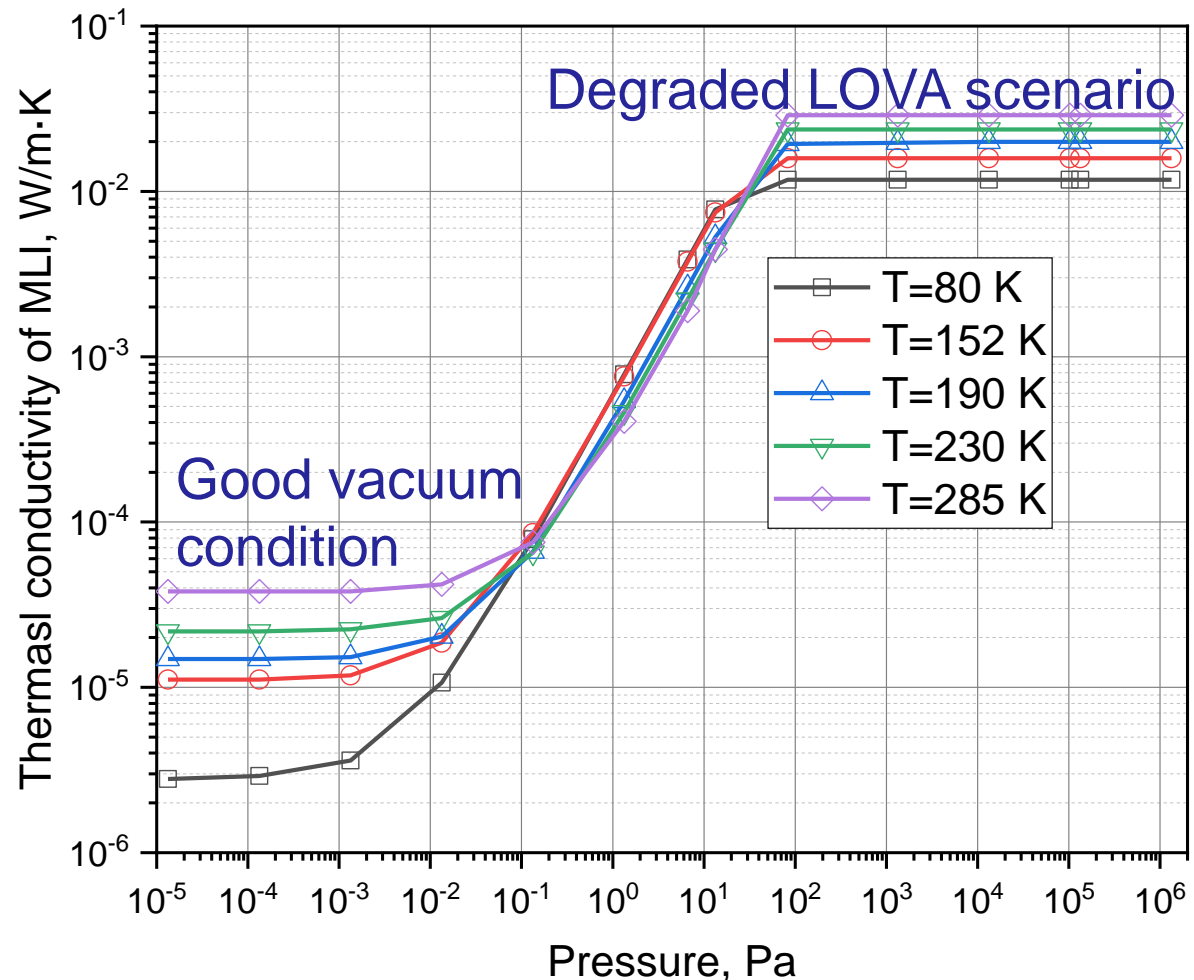
❑ Effective conductivity of MLI:

$$k_{ins} = k(P,T)$$

❑ $Cp_{ins} = 10 \text{ J}/(\text{kg}\cdot\text{K})$

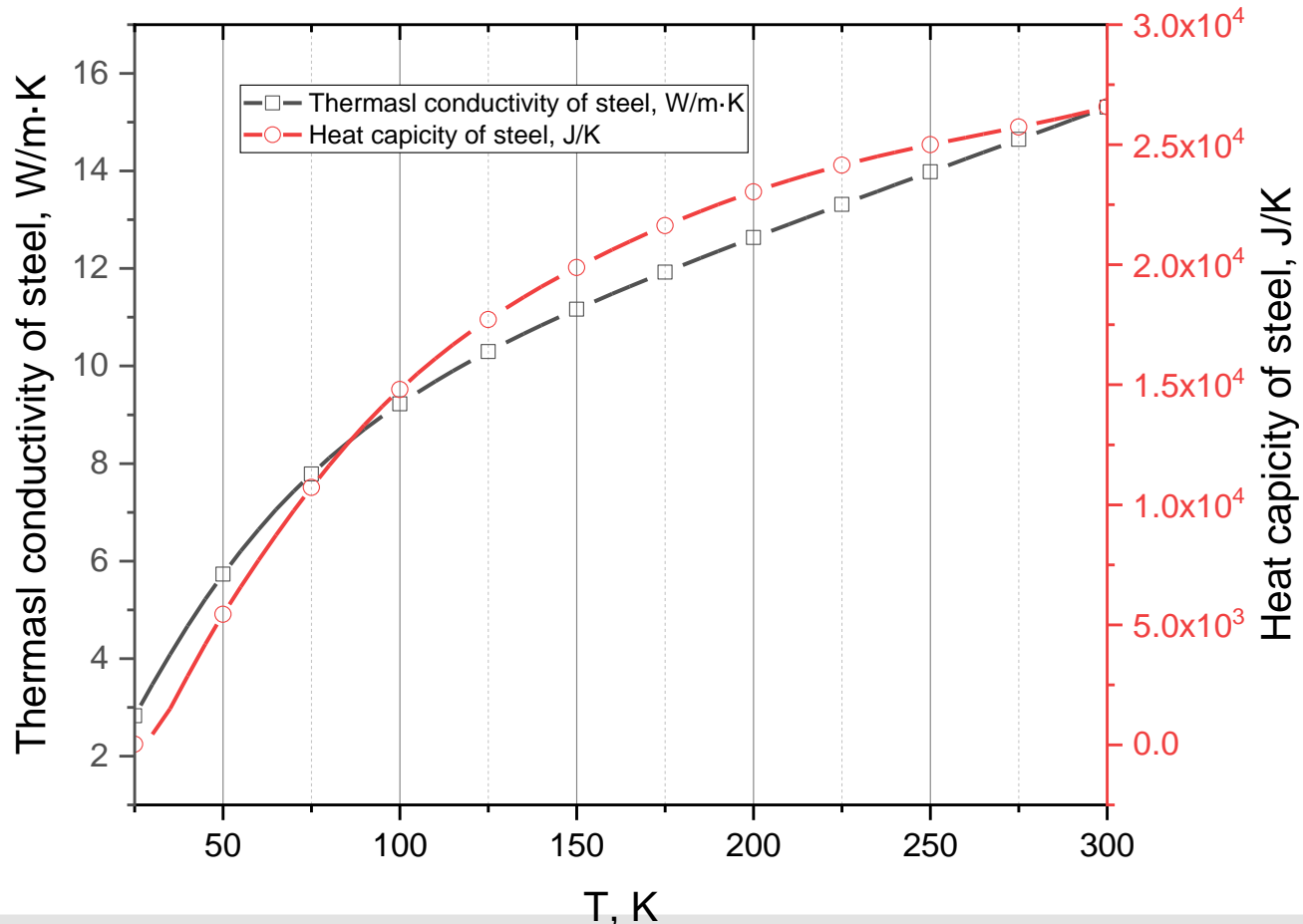
❑ Thickness $th_{ins} = 0.01 \text{ m}$

❑ Total mass $M_{ins} = 0.5 \text{ kg}$



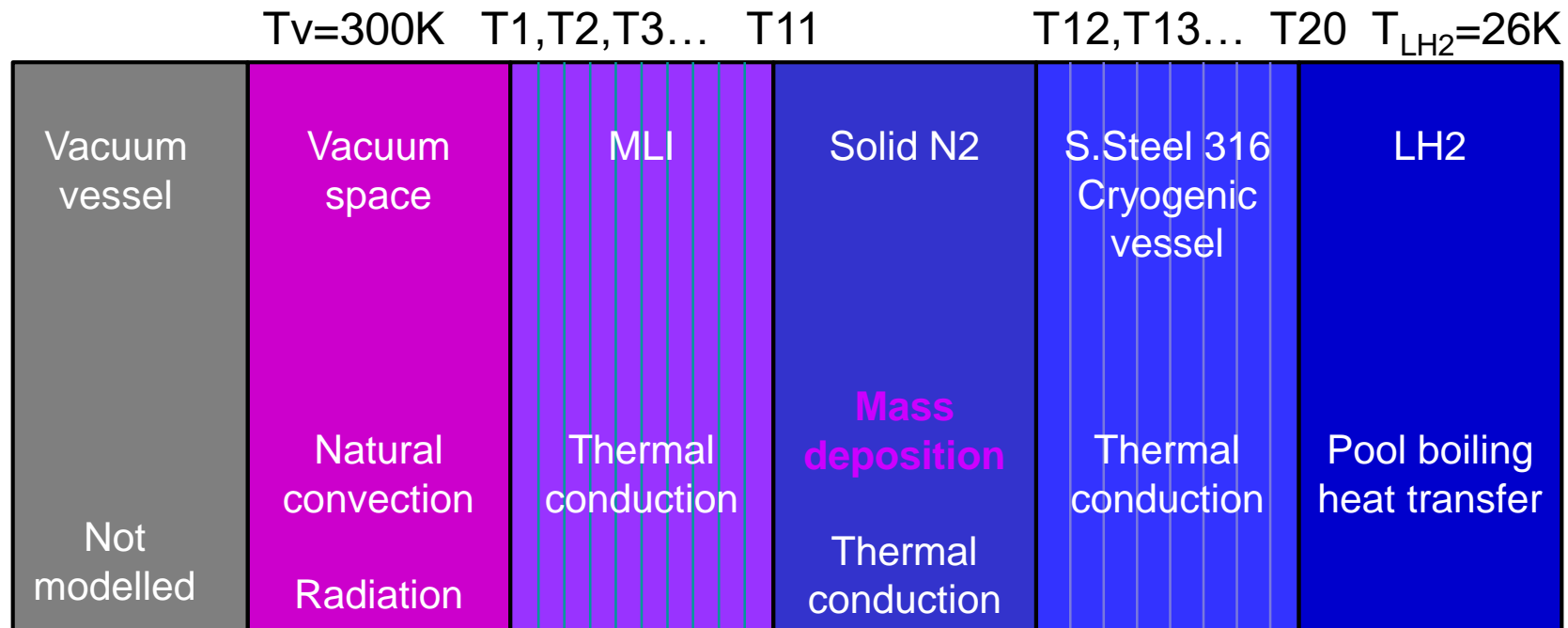
Material properties – S.Steel 316

- ❑ Thermal conductivity (k) and heat capacity (C_p) in 4K-300K from NIST
- ❑ Polynomial fitted



Lumped parameter model

□ Lumped parameter model for heat and mass transfer calculation



Dynamic venting flow at break

- Venting gas flow model: sonic or subsonic flow at vacuum break

$$m\dot{V}_v(P_v) = \begin{cases} A_v(2P_a\rho_a)^{0.5} \left(\frac{2}{\kappa_a + 1}\right)^{\frac{1}{\kappa_a - 1}} \left(\frac{\kappa_a}{\kappa_a + 1}\right)^{0.5} & , \text{if } P_v < P_{vc} \\ A_v(2P_a\rho_a)^{0.5} \left\{ \frac{\kappa_a}{\kappa_a - 1} \left[\left(\frac{P_v}{P_a}\right)^{\frac{2}{\kappa_a}} - \left(\frac{P_v}{P_a}\right)^{\frac{\kappa_a + 1}{\kappa_a}} \right] \right\}^{0.5} & , \text{if } P_v \geq P_{vc} \end{cases}$$

$$P_{vc} = P_a \left(\frac{2}{\kappa_a + 1}\right)^{\frac{\kappa_a}{\kappa_a - 1}}$$

- Loss of vacuum

$$\frac{dP_v}{dt} = \frac{T_v R_a}{V_v} [m\dot{V}_v(P_v) - A_{cr} m_{dep}(P_v, T_v, T_{cr})]$$

□ Mass deposition

$$m_{dep}(P_v, T_v, T_{cr}) = k_{dep} [\rho_v(P_v, T_v) - \rho_{sat}(T_{cr})]$$

$$k_{dep} = 0.075 \text{ m/s}$$

$$q_{dep}(P_v, T_v, T_{cr}) = m_{dep}(P_v, T_v, T_{cr}) [h_v(T_v) - h_{cr}(T_{cr})]$$

$$\frac{dM_{sn}}{dt} = m_{dep} A_{sn}$$

Convection model

□ Natural convection in vacuum vessel

□ Based on natural convection between two vertical parallel plates

$$Nu = \begin{cases} 0.197Ra^{0.25} \left(\frac{L}{H}\right)^{\frac{1}{9}}, & \text{if } Ra < 2 \cdot 10^5 \\ 0.073Ra^{\frac{1}{3}} \left(\frac{L}{H}\right)^{\frac{1}{9}}, & \text{if } Ra \geq 2 \cdot 10^5 \end{cases}$$

$$\alpha = \frac{Nu \cdot k_{N2}}{L}$$

$$q_{conv} = \alpha(T_v - T_1)$$

- **G.F. Xie, et.al. Study on the heat transfer of high-vacuum-multilayer-insulation tank after sudden, catastrophic loss of insulating vacuum, *Cryogenics* 50 (2010) 682–687**

Radiation model

□ Thermal radiation in vacuum vessel

□ Based on radiation model between two concentric cylinders

$$q_{rad} = \frac{\sigma(T_v^4 - T_1^4)}{\frac{1}{\epsilon_{ss}} + \frac{A_{ss}}{A_{al}} \left(\frac{1}{\epsilon_{al}} - 1 \right)}$$

$$\epsilon_{ss} := 0.8$$

$$\epsilon_{alu} := 0.15$$

- Weber C, Henriques A and Grohmann S, Study on the heat transfer of helium cryostats following loss of insulating vacuum, IOP Conf. Series: Materials Science and Engineering 502 (2019) 012170, doi:10.1088/1757-899X/502/1/012170

Pool boiling heat transfer model

□ Pool boiling heat transfer

□ Regime of nucleate pool boiling, due to heat flux $\sim 1-2 \text{ W/cm}^2$ at cryogenic condition

□ Kutateladze (1952) correlation, (improved by Rohsenow, 1952)

$$q_{boil}(T_w) = \mu_f H_{fg} \left[\frac{g(\rho_f - \rho_g)}{\sigma_f} \right]^{0.5} \left[\frac{Kp^{0.7} C p_f (T_w - T_s)}{881 Pr_f^{0.65} H_{fg}} \right]^{\frac{10}{3}}$$

$$Kp = \frac{\rho_f P_s}{\rho_g [g \sigma_f (\rho_f - \rho_g)]^{0.5}}$$

“The pool boiling correlation of Kutateladze (Kutateladze 1952; Brentari and Smith 1965) has been verified for cryogenic fluids including N₂, O₂, H₂, and He”, cited from

- **Cryogenic Heat Transfer, Second Edition, by R F Barron**

□ Thermal conduction through MLI

$$R_{ins}(P_v, T_{ins}) = \frac{th_{ins}}{k_{ins}(P_v, T_{ins})}$$

□ Thermal conduction through iced layer

$$R_{sn}(M_{sn}, T_{sn}) = \frac{th_{sn}(M_{sn}, T_{sn})}{k_{sn}(T_{sn})} \quad th_{sn}(M_{sn}, T_{sn}) = \frac{M_{sn}}{\rho_{sn}(T_{sn})A_{sn}}$$

□ Thermal conduction through vessel wall (S.S.316)

$$R_{ss}(T_{ss}) = \frac{th_{ss}}{k_{ss}(T_{ss})}$$

Governing equation group (ODEs)

T1, T2, T3... T11

T12, T13... T20

22 independent variables
22 ODEs

Vacuum (P_v)

MLI

Solid N₂ (M_{sn})

S.Steel 316

LH2

$$\frac{d}{dt} P_v = \frac{T_v \cdot Ra}{V_{vacuum}} \cdot (m \ln(P_v) - Acr \cdot m_{dep}(P_v, T_{11}))$$

State equation of gas in vacuum closure

$$\frac{d}{dt} M_{sn} = A_{sn} \left(M_{sn}, \frac{T_{11} + T_{12}}{2} \right) \cdot m_{dep}(P_v, T_{11})$$

Mass deposition rate equation

Fourier law based heat transfer equations

$$\frac{d}{dt} \left(\frac{T_i + T_{i+1}}{2} \right) = \frac{A_i}{M_i \cdot Cp_i} \left[q_{src} + \frac{T_{i-1} - T_i}{R_{i-1} \left(\frac{T_{i-1} + T_i}{2} \right)} - \frac{T_i - T_{i+2}}{R_i \left(\frac{T_i + T_{i+1}}{2} \right) + R_{i+1} \left(\frac{T_{i+1} + T_{i+2}}{2} \right)} \right] \quad (i = 1, 2 \dots 19)$$

Boiling heat transfer on cryogenic vessel wall

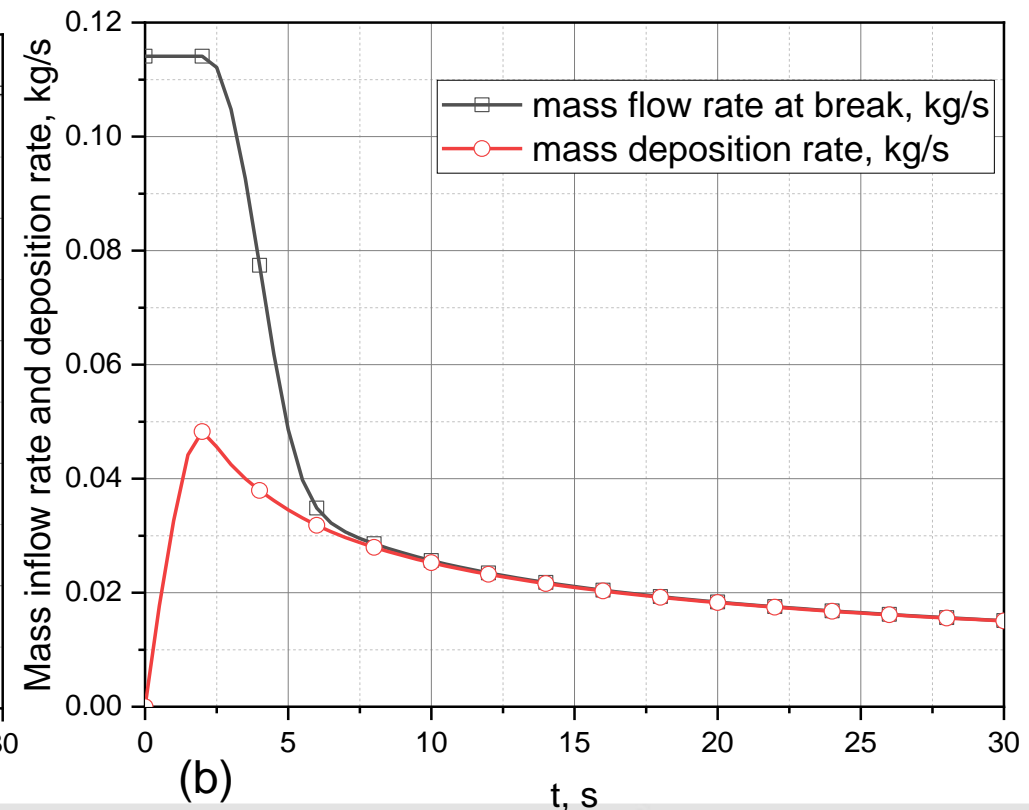
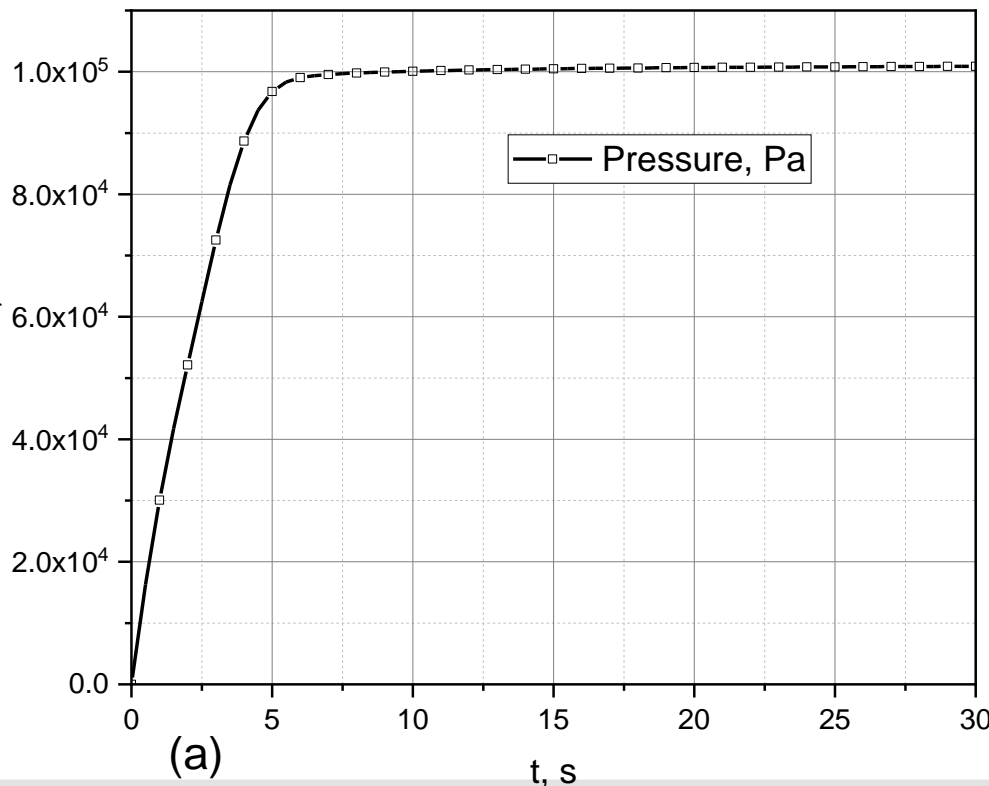
$$\frac{dT_i}{dt} = \frac{A_i}{M_i \cdot Cp_i} \left[\frac{T_{i-1} - T_i}{R_{i-1} \left(\frac{T_{i-1} + T_i}{2} \right)} - q_{boil} \right] \quad (i = 20)$$

Calculation results

□ Initial conditions Pa = 1 atm. Ta = 300 K D = 0.025 m

□ Pressure in vacuum closure

□ Venting gas mass flow rate and deposition rate

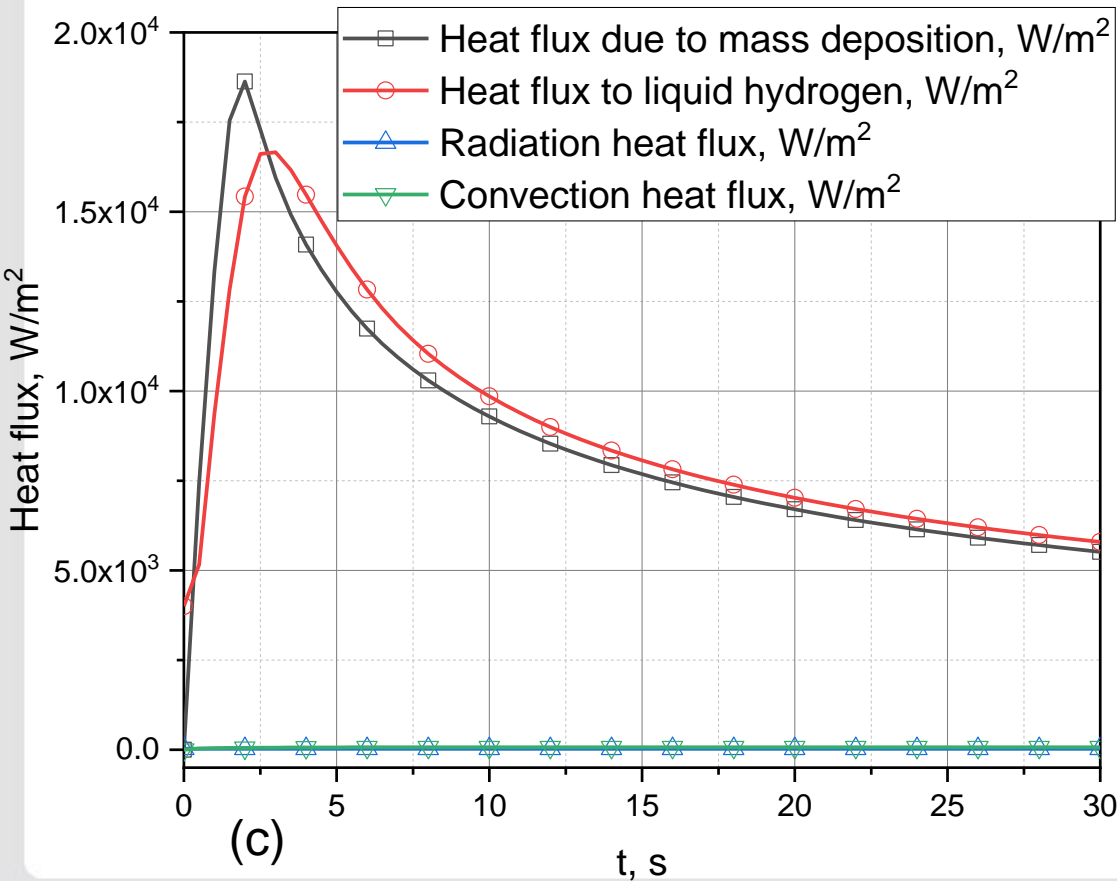


Calculation results

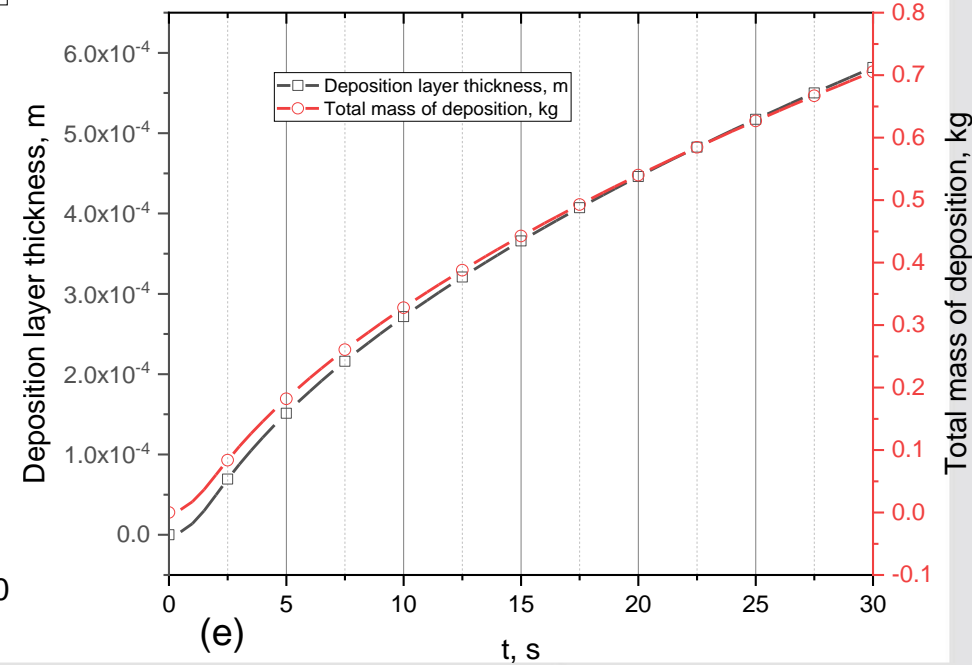
Heat fluxes due to mass deposition

Max. heat flux: **1.87 W/cm²**

Max. heat flux to LH2: **1.67 W/cm²**



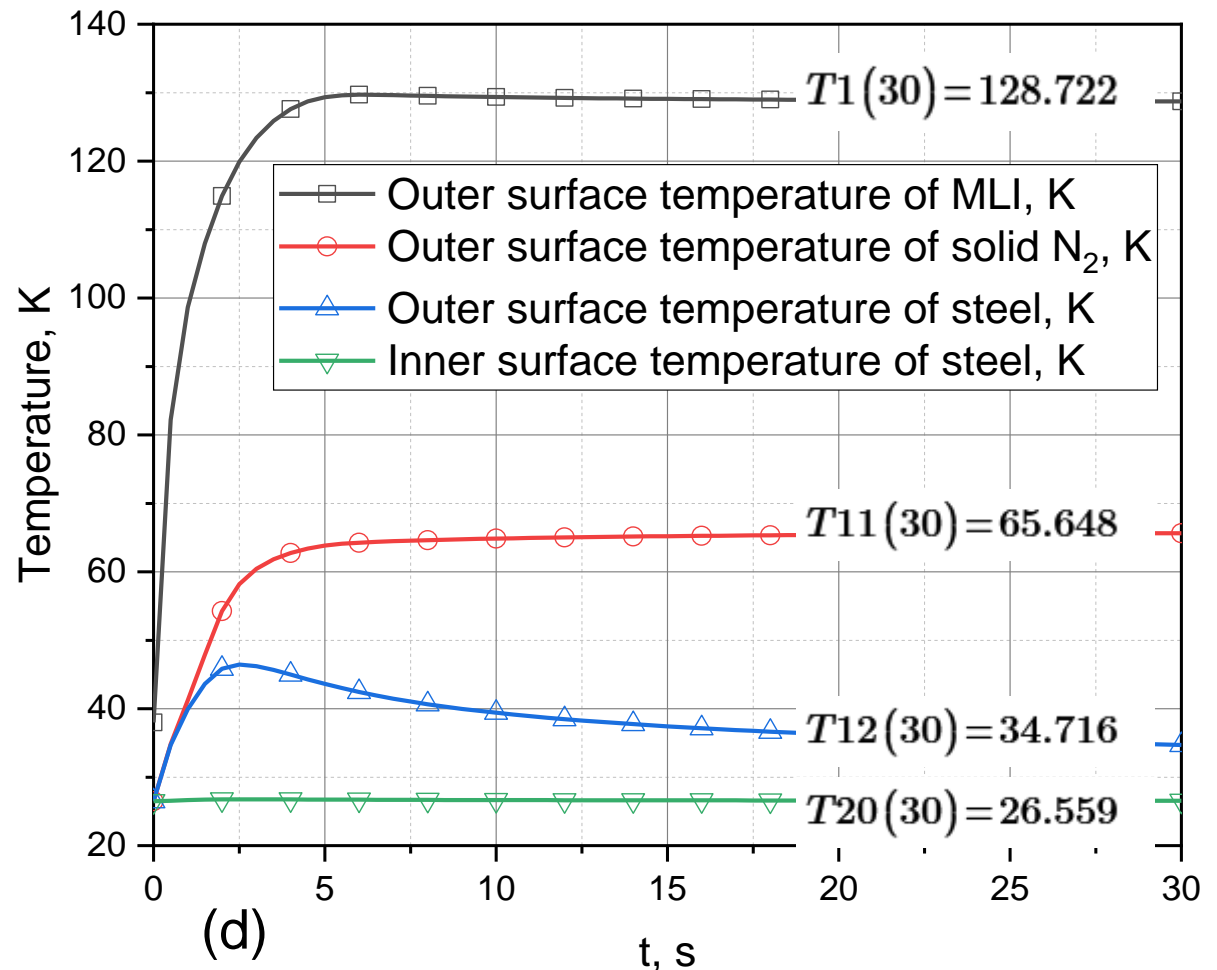
Iced N2 layer thickness and mass



Calculation results

- Temperatures at surfaces of MLI (T1), iced layer (T11), outer surface of steel (T12) and inner surface of steel (T20)

- $\Delta T_{\text{MLI}} = 63 \text{ K}$
- $\Delta T_{\text{solid N}_2} = 31 \text{ K}$
- $\Delta T_{\text{steel}} = 8 \text{ K}$
- $\Delta T_{\text{boiling}} = 0.6 \text{ K}$



Conclusions

- ❑ Material properties are precise as possible, including gaseous, liquid and solid nitrogen, gaseous and liquid hydrogen, and stainless steel 316 at cryogenic condition
- ❑ Lump parameter models of heat and mass transfer are developed to find heat load to LH2 in LOVA scenario, by modeling MLI, iced layer, convection on venting side and pool boiling heat transfer on LH2 side
- ❑ Mass deposition of venting gas dominates heat load to LH2, thermal radiation and convection play significantly less roles
- ❑ Venting mass does not deposit on outer surface of MLI, but in region between MLI and cryogenic wall if temperature is low enough
- ❑ In case of $\Phi 2.5$ cm break, maximal heat load to LH2 is ~ 1.67 W/cm², iced layer thickness $\sim \frac{1}{2}$ mm in the first 30 s in LOVA
- ❑ Thermal resistance of $\frac{1}{2}$ mm thick iced layer is not ignorable, due to its low conductivity at cryogenic temperature. The resulted temperature decrease (31K) is about half of that of MLI (63K)
- ❑ Verification against experimental data is expected next step