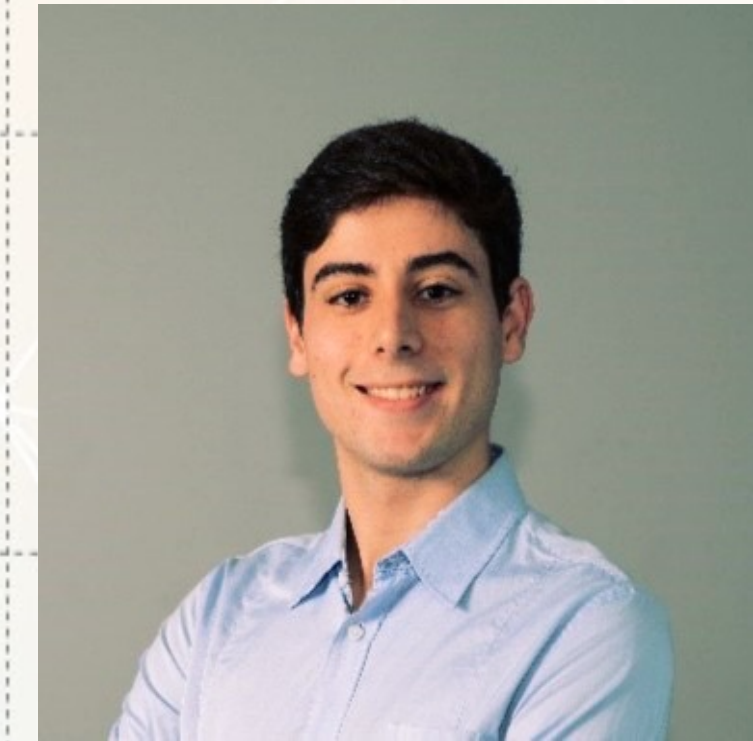




CENTRE EUROPÉEN DE RECHERCHE ET DE FORMATION AVANCÉE EN **CALCUL SCIENTIFIQUE**



B. Vanbersel



F. Meziat

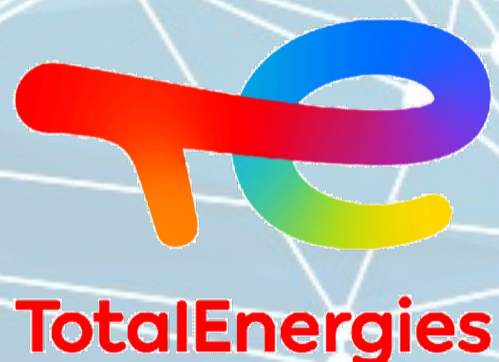
LARGE EDDY SIMULATIONS OF A HYDROGEN-AIR EXPLOSION IN AN OBSTRUCTED CHAMBER USING ADAPTIVE MESH REFINEMENT

Vanbersel, B.¹, Meziat Ramirez, F.A.^{1,2}, Vermorel, O.¹, Jaravel, T.¹, Douasbin, Q.¹ and **Dounia, O.¹**

¹ CERFACS, CFD Team, Toulouse, Cedex 01 31057, France, vanbersel@cerfacs.fr

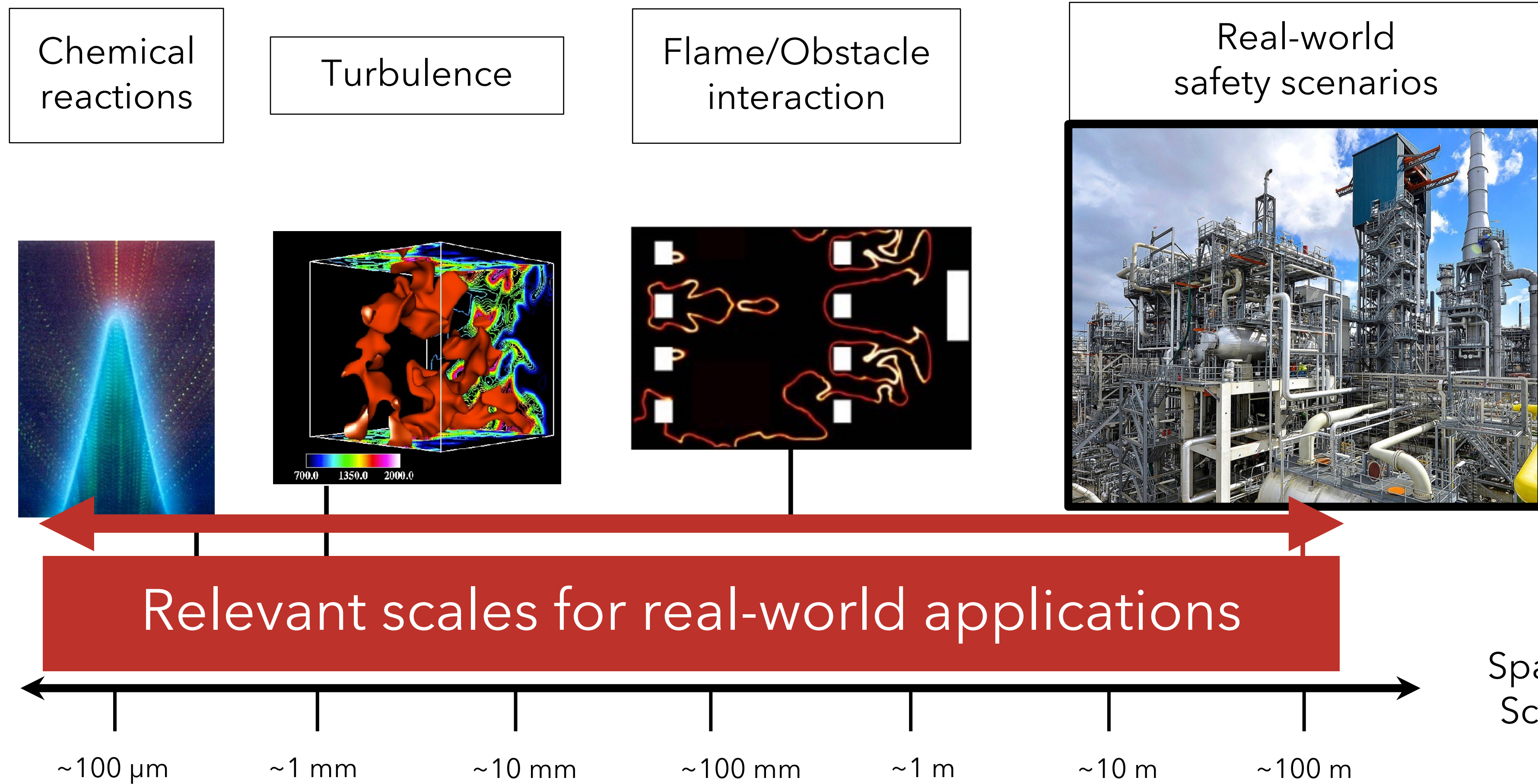
² Air Liquide – Paris Innovation Campus, Les Loges-en-Josas, 78354, France

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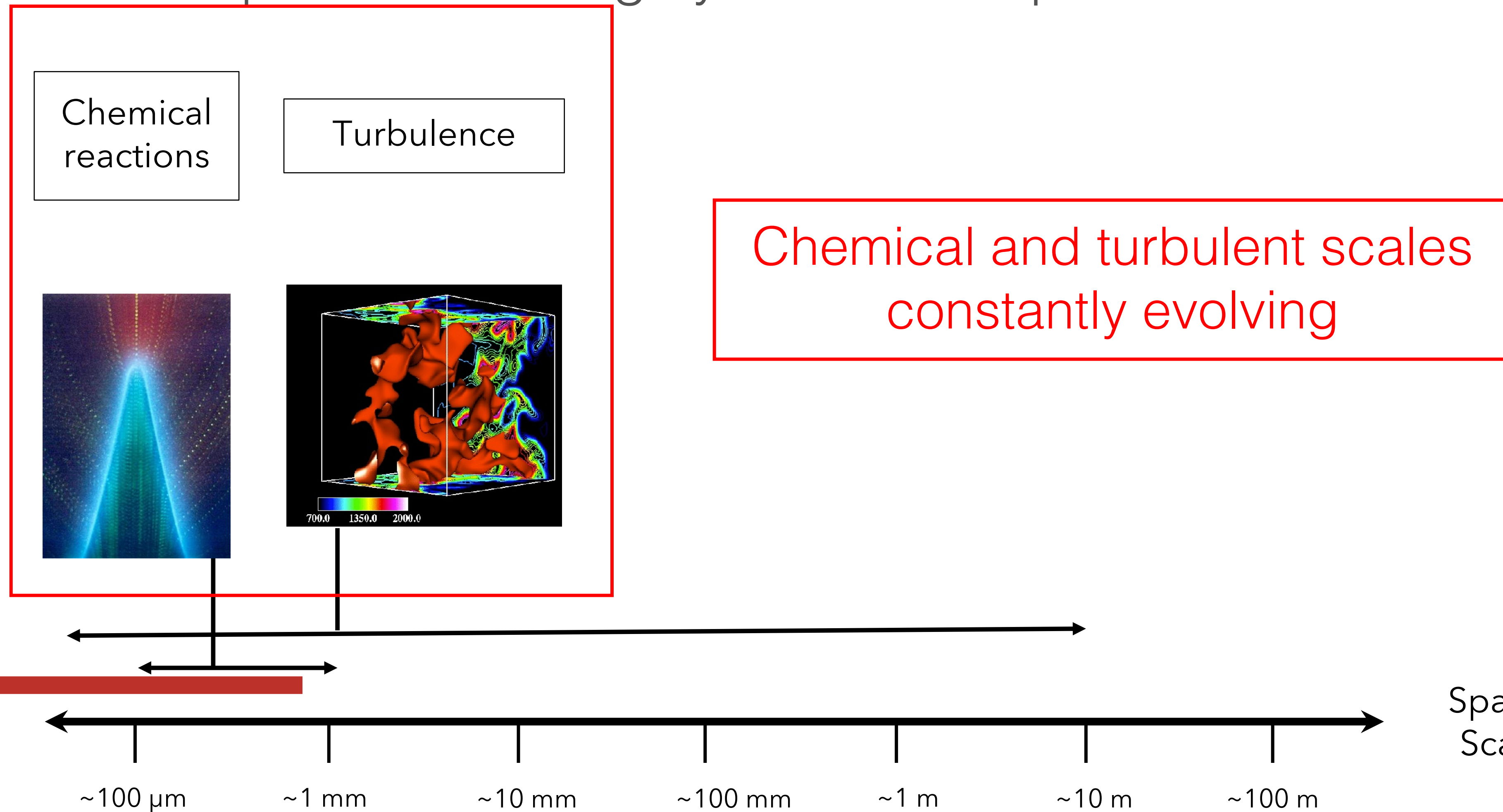
CFD for Explosions: the main challenge

Explosions are highly multi-scale phenomena



Explosion physics: the main challenge

Explosions are highly multi-scale phenomena



LES for High-fidelity prediction of relatively large scale explosions

DNS

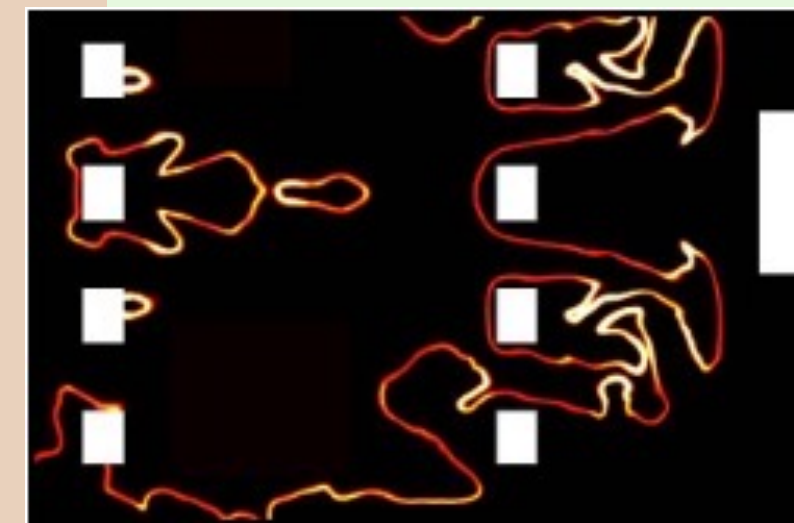
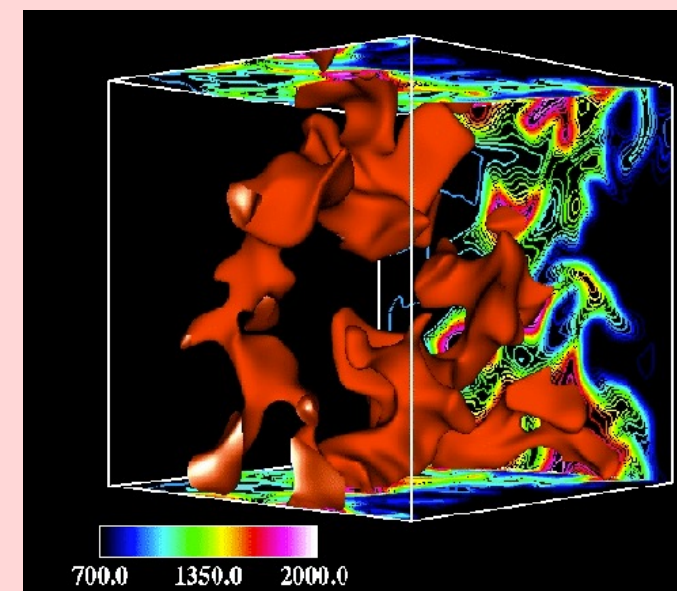
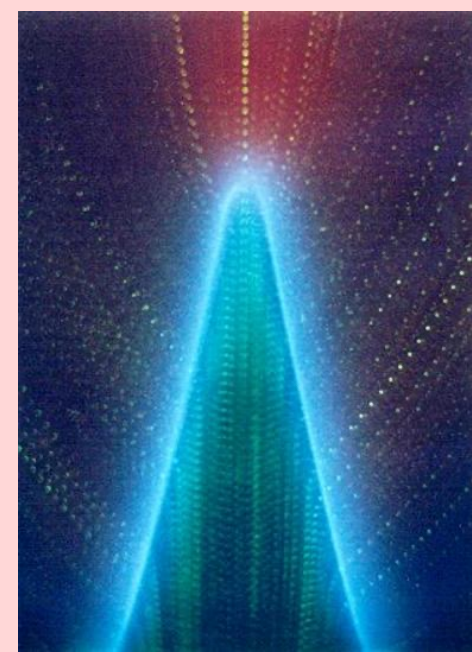
Subgrid-scale models

Large Eddy Simulation (LES) :

- Bridge the gap between DNS and URANS range of applications
- Towards industrial relevance

=> **need for a massively parallel and efficient LES code**

URANS



~100 μm

~1 mm

~10 mm

~100 mm

~1 m

~10 m

~100 m

Spatial Scale

LES solver: AVBP



Unstructured LES/DNS solver

- Unstructured (hex., tet., prisms, hybrid)
- Explicit in time
- Massively parallel

- Multi-phase (Lagrangian & Eulerian coupling)

- Chemistry: tabulated, reduced, detailed

- Other state-of-art numerical techniques

For explosion applications

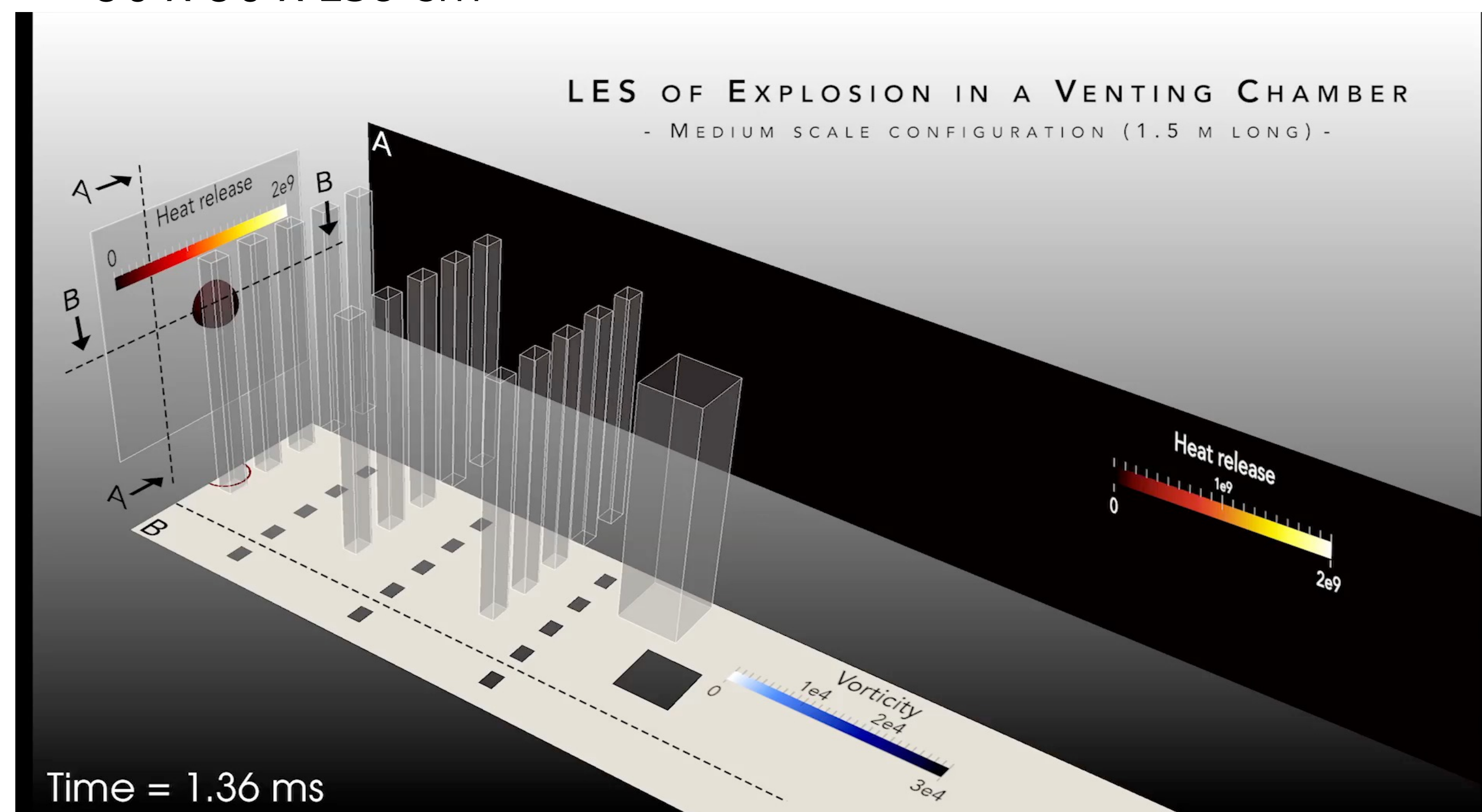
- **Numerics**
 - 3rd order (space and time)
 - artificial dissipation for shock handling
 - Dynamic Adaptive Mesh Refinement

- **LES Modeling**
 - Flow: Smagorinsky, WALE, Sigma, ...
 - Combustion:
 - *Thickened flame model (TFLES)*
 - Colin/Charlette efficiency functions

Operating point

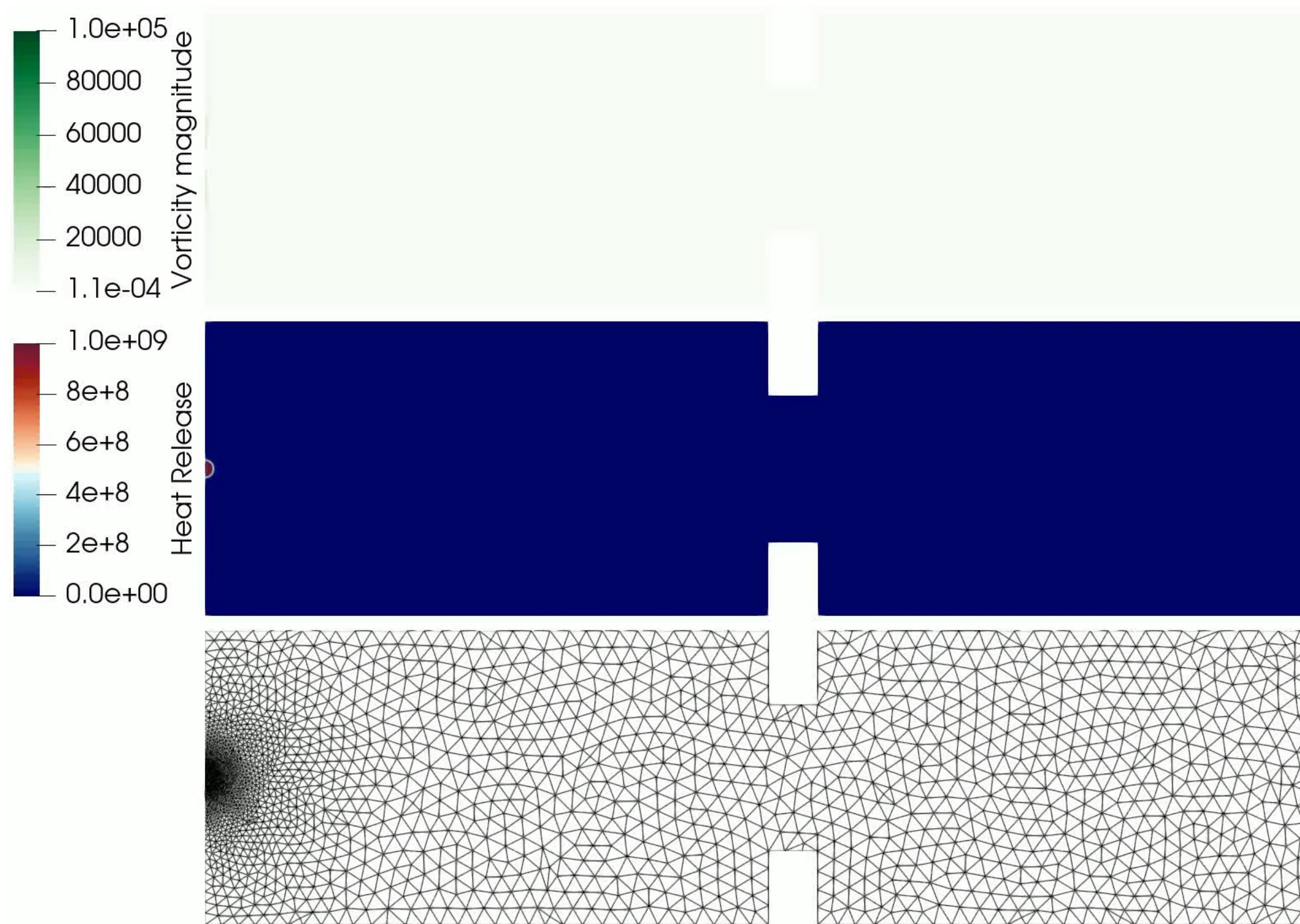
- 30 x 30 x **150** cm

- Premixed mixture (C₃H₈/Air)
- No initial turbulence
- Equivalence ratio: $\Phi = 1$
- One central square obstruction
- Turbulence generating grids
- Laser ignition at the closed end



Combine LES with Dynamic Mesh Refinement for flame acceleration problems

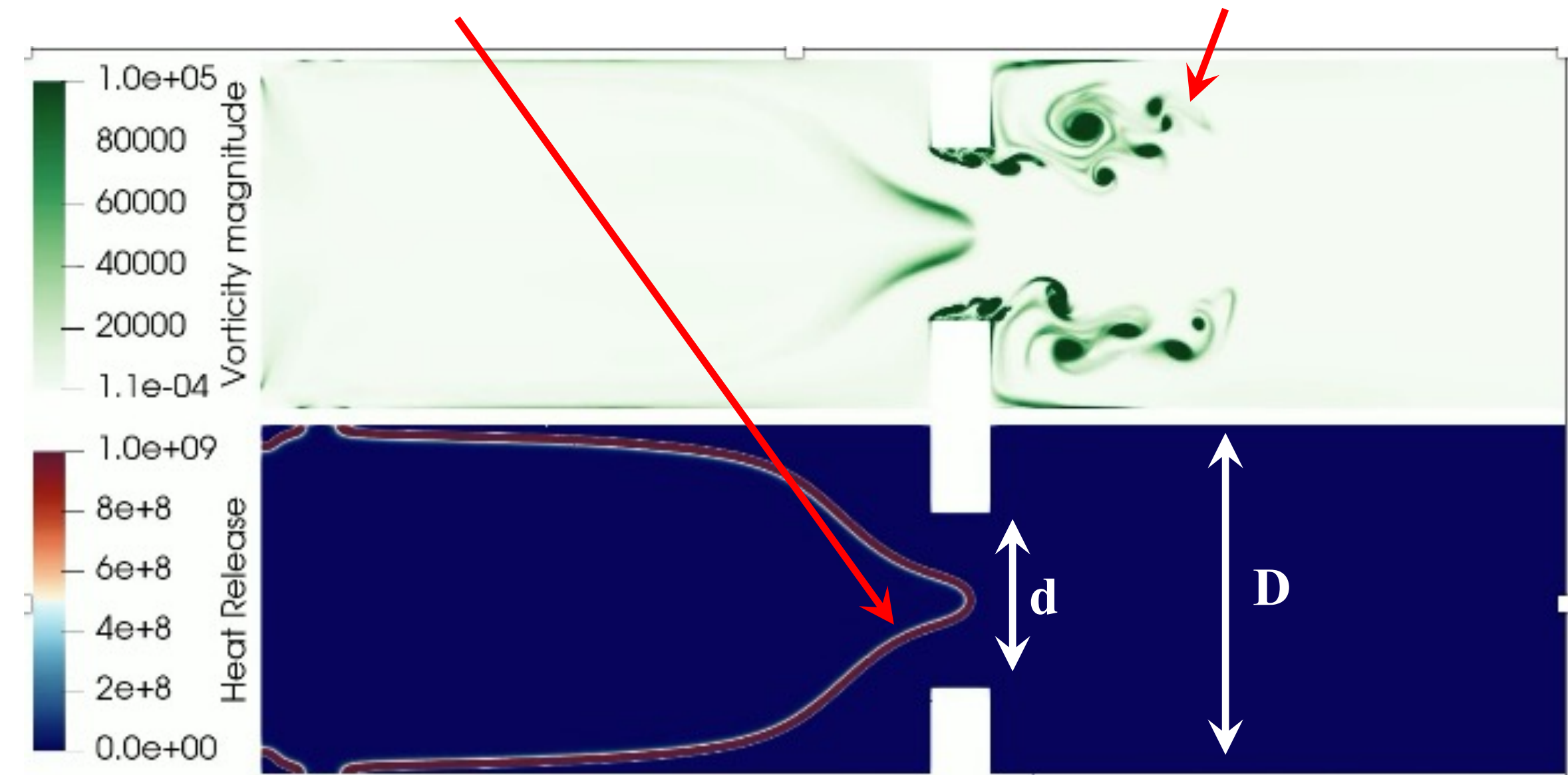
Flame/Obstacle interaction problem



Good resolution needed to accurately predict:

turbulent flame propagation

vortex shedding



⇒ Large scale problem + Static mesh ⇒ Not affordable

⇒ Combine LES with Dynamic Mesh Refinement

AMR library for **unstructured grids**:

YALES2[1]/Kalpataru(CERFACS) + MMG3D [2]

[1] P. Benard, et al, *International Journal for Numerical Methods in Fluids* **81**, 719 (2016).

[2] C. Dapogny, et al, *Journal of Computational Physics* **262**, 358 (2014).



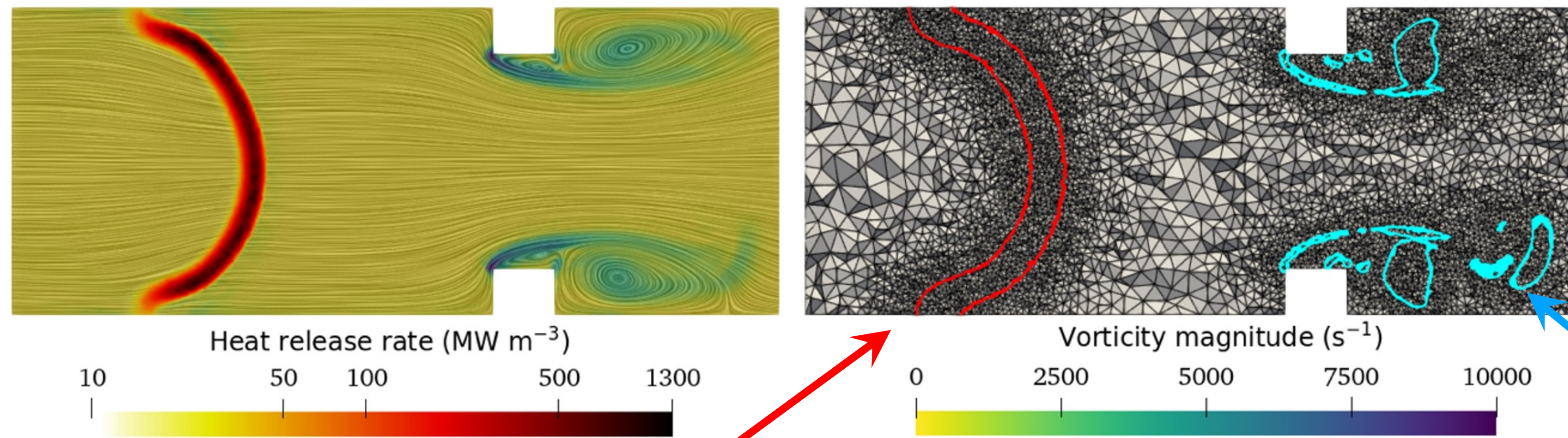
Objectives:

- **Develop an AMR strategy for LES of fast deflagrations in large scale configurations (+ unstructured grids constraint)**
- **Get rid (AMAP) of user defined parameters**

1. AMR recipe for high-fidelity LES of gas explosions

2. Validation on a fast deflagration case (Gravent)

AMR recipe for high-fidelity LES [1/2]



Flame

Vortex shedding

Regions of interest:

- **Metric** : θ_F ^[7] (from the DTFLES model)
- **Resolution**: Δx_1
 - $F \sim [10-30]$
 - **Relatively thin flame**: $F\delta_L \ll d$

- **Metric** : Ω ^[8] to detect regions where
 - **rotation overtakes deformation**
- **Resolution**: Δx_1
 - **all regions of interest share the same Δx_1** ^[9]

Outside Regions of interest:

- Sufficient resolution to capture:**
- **Flow contraction**
 - **flow profiles in un-obstructed regions**

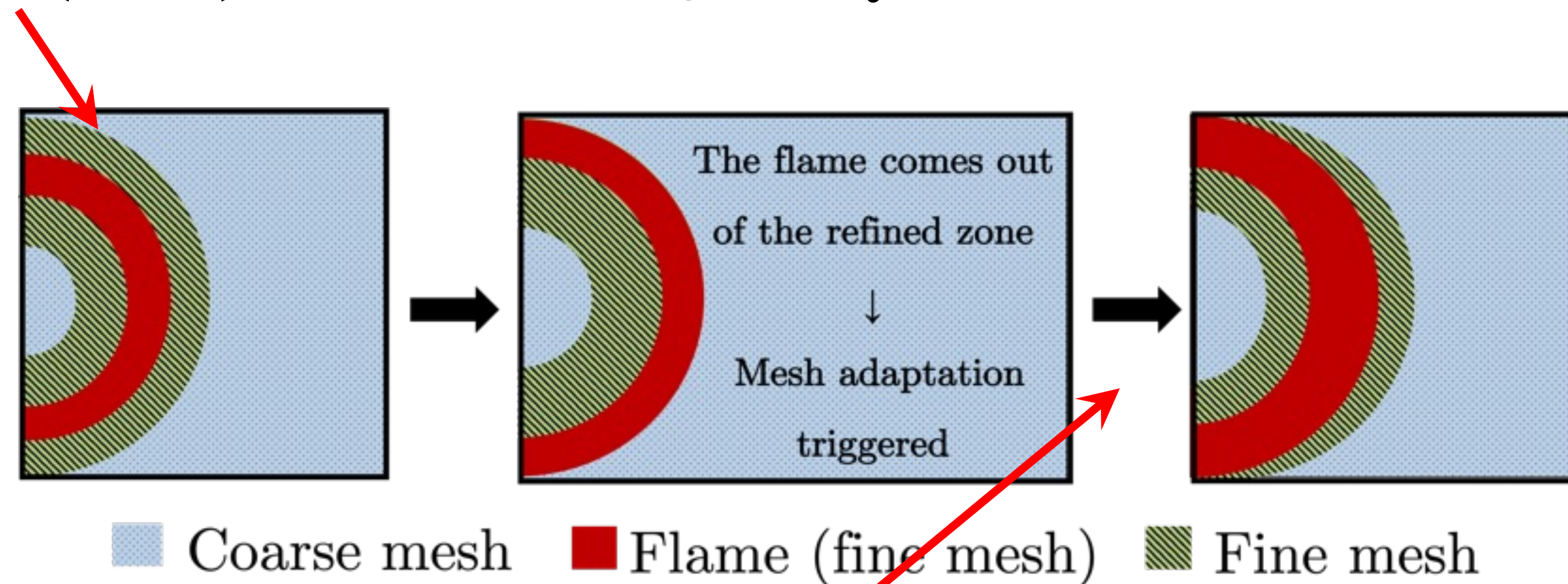
[7] J. P. Legier et al., *Proceedings of the Summer Program, CTR, 2000.*

[9] C. Mehl, S. Liu, O. Colin, FTACS, 2021

[8] C. Liu et al., *Science China: Physics, Mechanics and Astronomy, 2016*

AMR recipe for high-fidelity LES [2/2]

Mask dilatation (1 cell) : ensures the QoI stays inside the RoI



Automatic mesh triggering:

- **when the flame reaches the limit of the RoI**
- **No user-defined parameter**

QoI: Quantity of Interest
RoI: Region of Interest

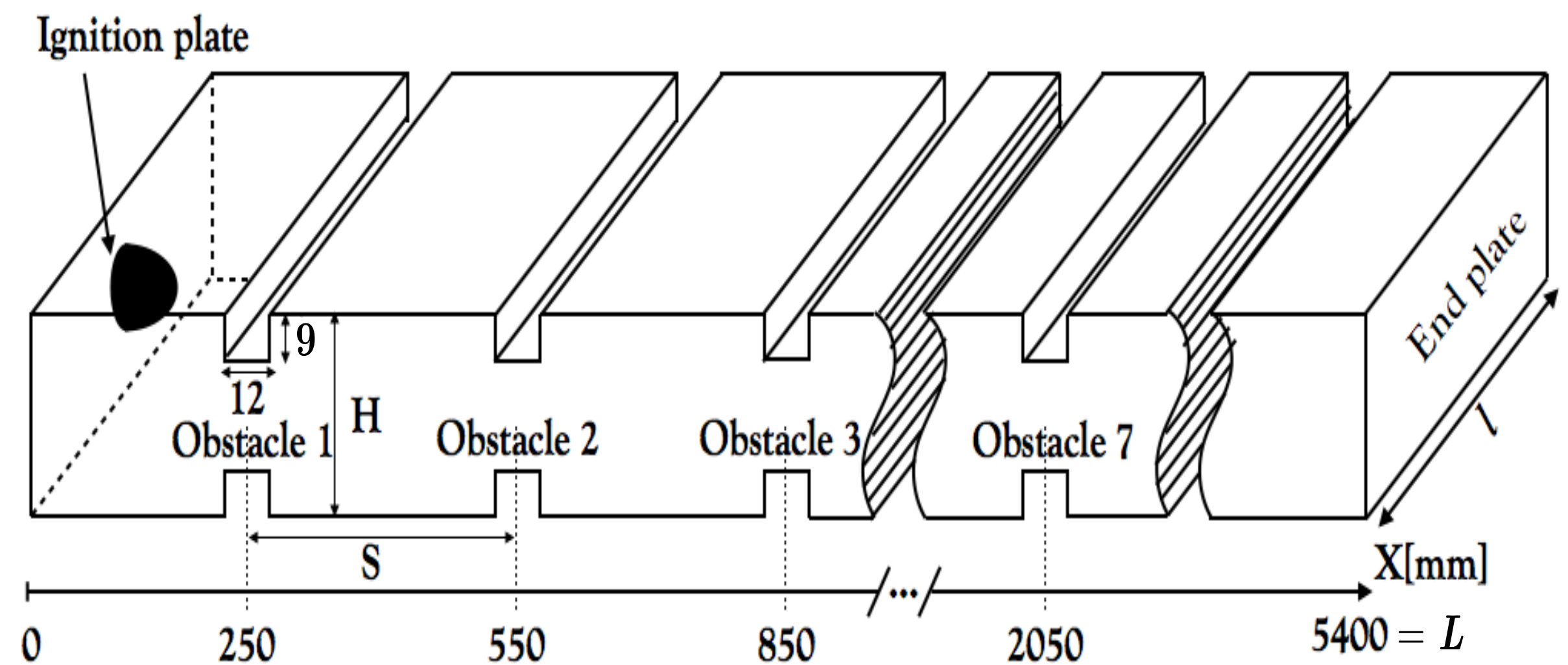


1. AMR recipe for high-fidelity LES of gas explosions

2. Validation on a fast deflagration case (Gravent)

Gravent BR30hS300 Test case^[2,4]

- Entirely closed rectangular channel (5.4 (L) x 0.3 (l) x 0.06 m (H)),
- Seven obstacles (BR=30%, S=300mm)
- H₂-air perfectly premixed: $\phi = 0.52$, $P = 1\text{atm}$, $T = 300\text{K}$
- Validation of the AMR strategy: Focus on Flame acceleration



GraVent BR30hS300 configuration^[3]

[2] L. Boeck et al., The gravent ddt database, Shock Waves 26 (2016).

[3] O. Dounia, Ph.D. thesis, INP Toulouse (2018).

[4] L. Boeck, Ph.D. thesis, Technische Universität München (2015).

Numerical setup

➤ LW convection scheme[4]: 2nd order in space and time.

➤ Isothermal Walls with law of the wall

➤ LES:

➤ Subgrid turbulent stresses: WALE [5]

➤ Flame thickening DTFLES

➤ Turbulent combustion: Colin [6]

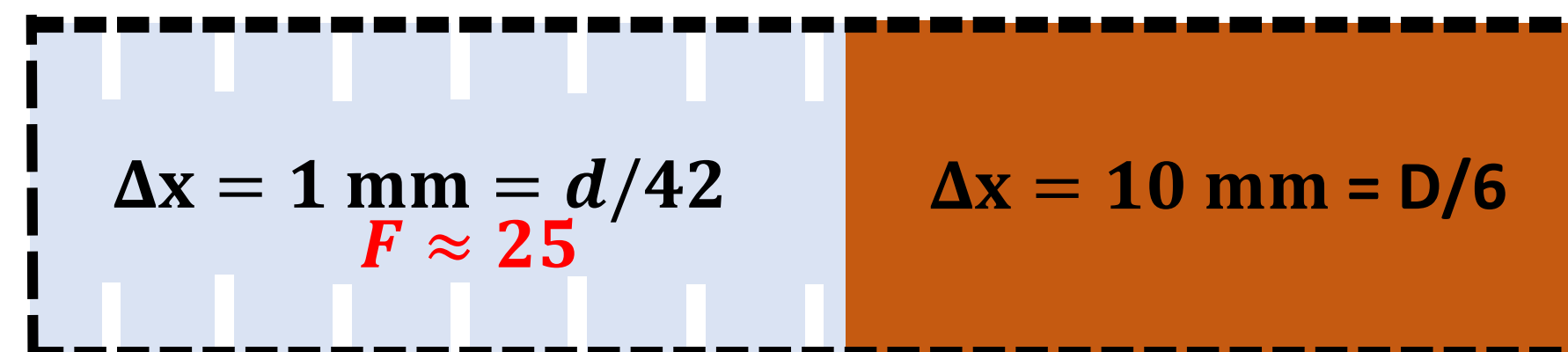
➤ **Meshes: fully unstructured tetrahedral meshes**

➤ Chemistry: $H_2 + \frac{1}{2}O_2 \rightarrow H_2O$

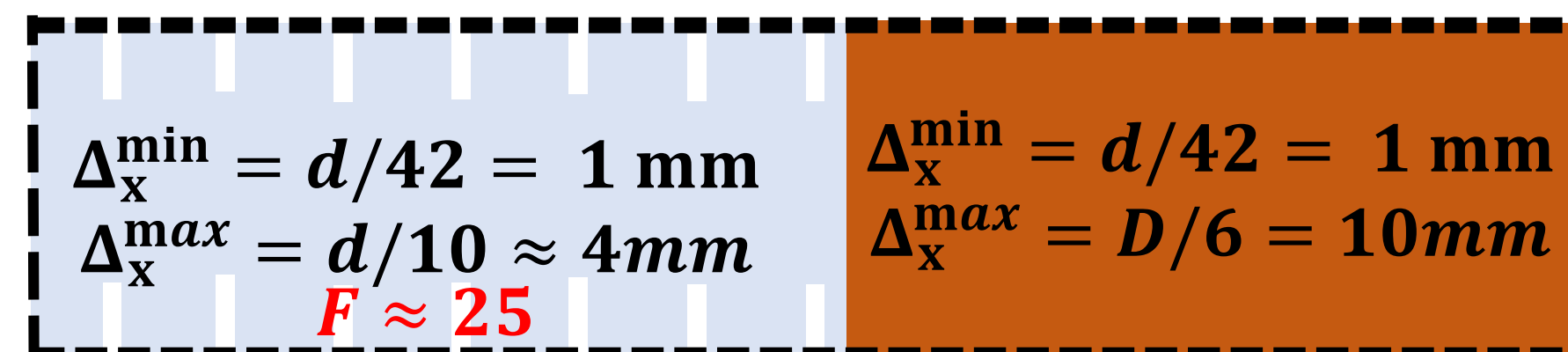
| ϕ | s_L^0 (ms ⁻¹) | T_{ad} (K) | δ_L^0 (m) |
|--------|-----------------------------|--------------|----------------------|
| 0.52 | 0.59 | 1657 | $2.03 \cdot 10^{-4}$ |

Table 1: Laminar H₂-air flame characteristics at $T_0 = 300$ K and $p_0 = 1$ atm for the single-step chemical scheme.

➤ Static REF (**380Mcells**):



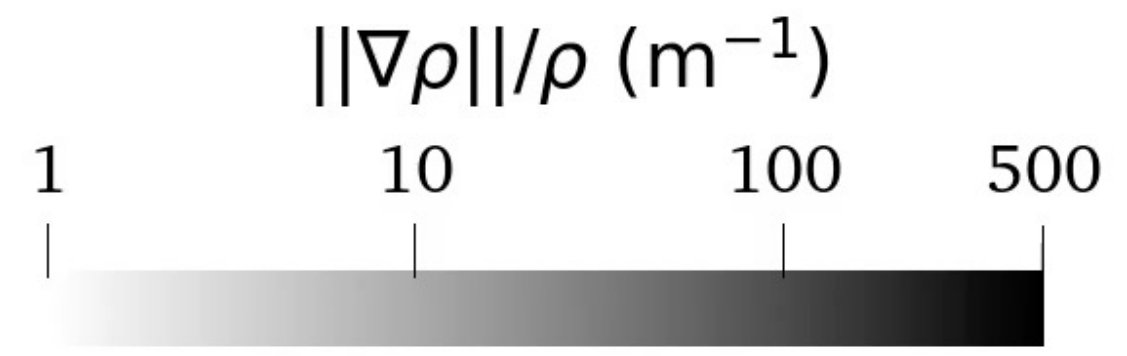
➤ Dynamic AMR



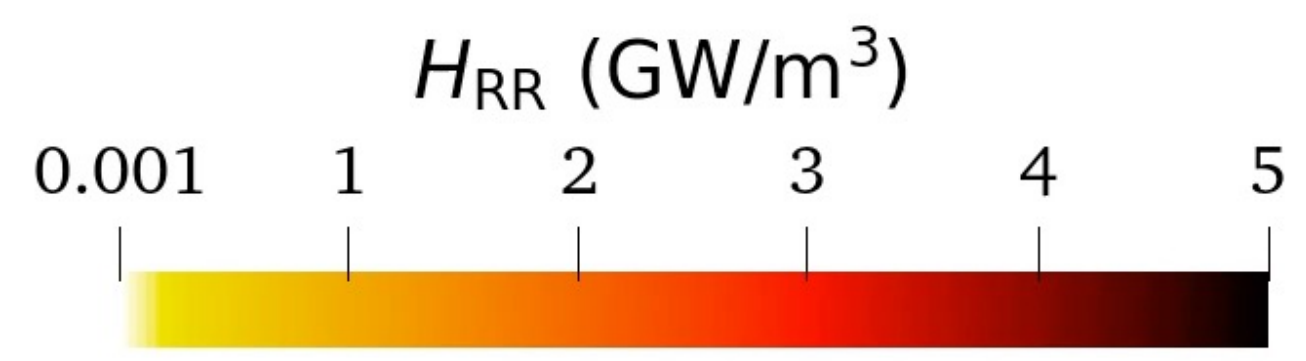
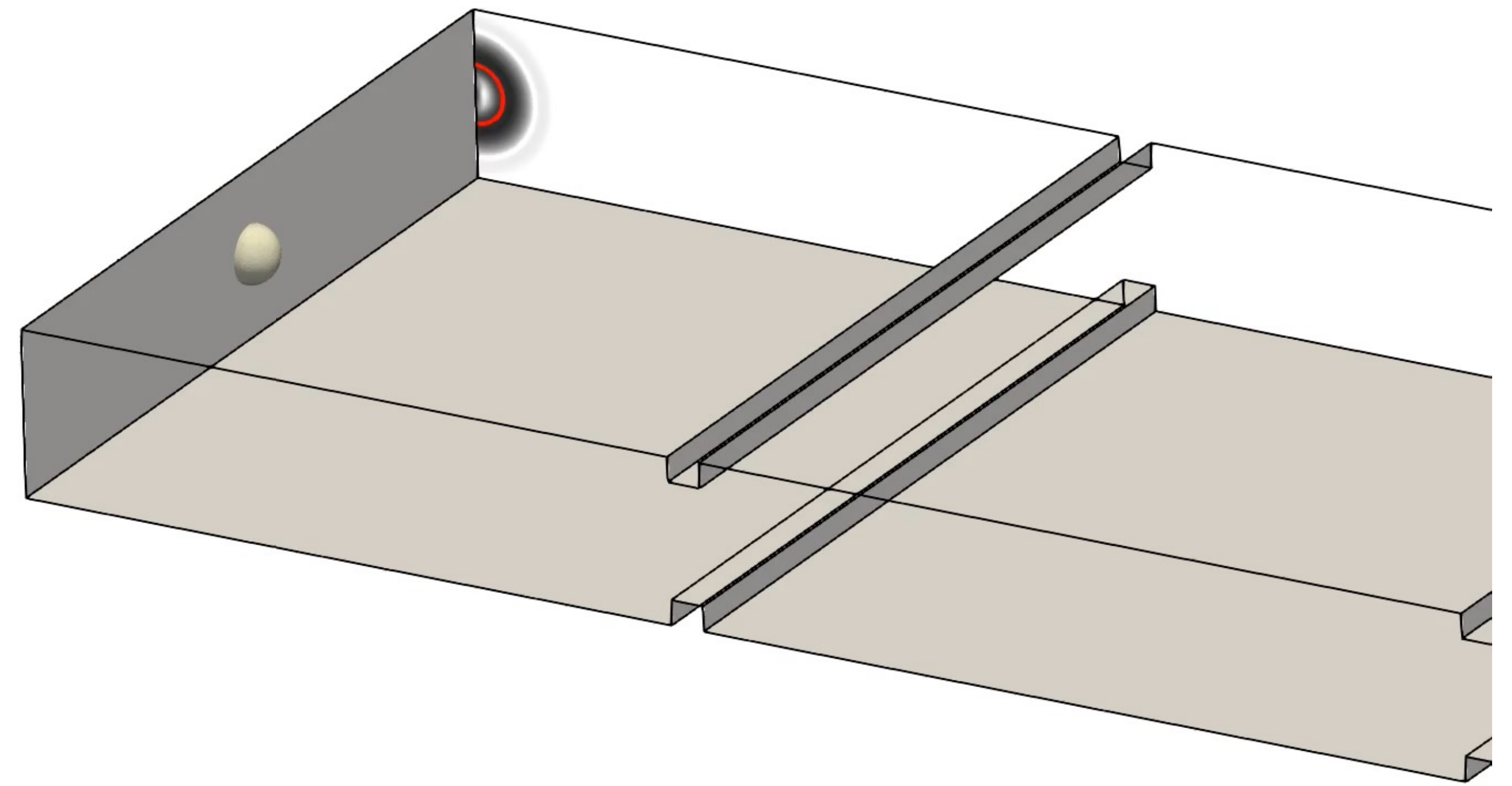
- [4] P. Lax and B. Wendroff, *Communications on Pure and Applied Mathematics* 1960.
- [5] F. Nicoud and F. Ducros, *Flow, Turbulence and Combustion* 1999.
- [6] O. Colin, et al., *Physics of Fluids*, 2000.



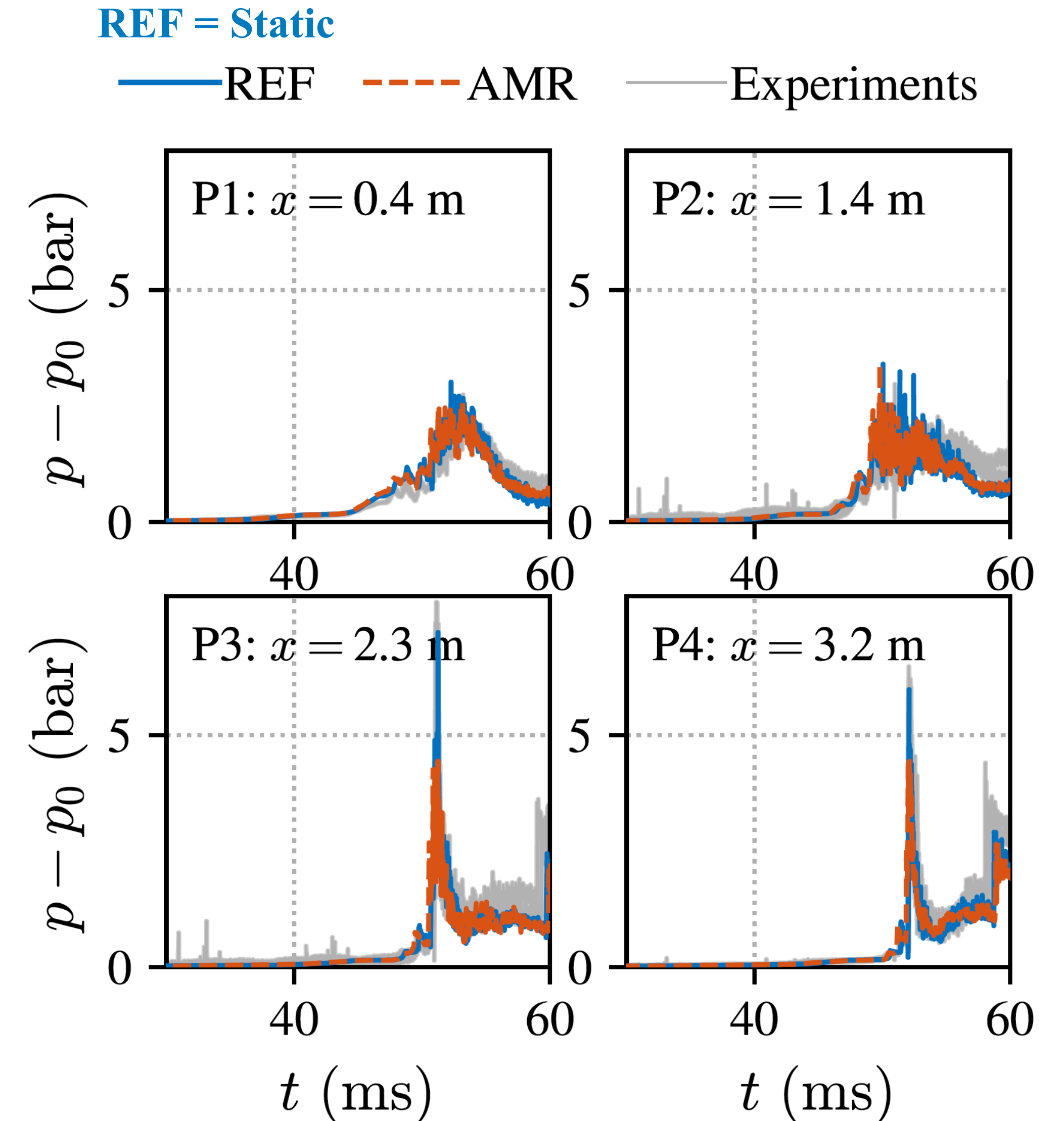
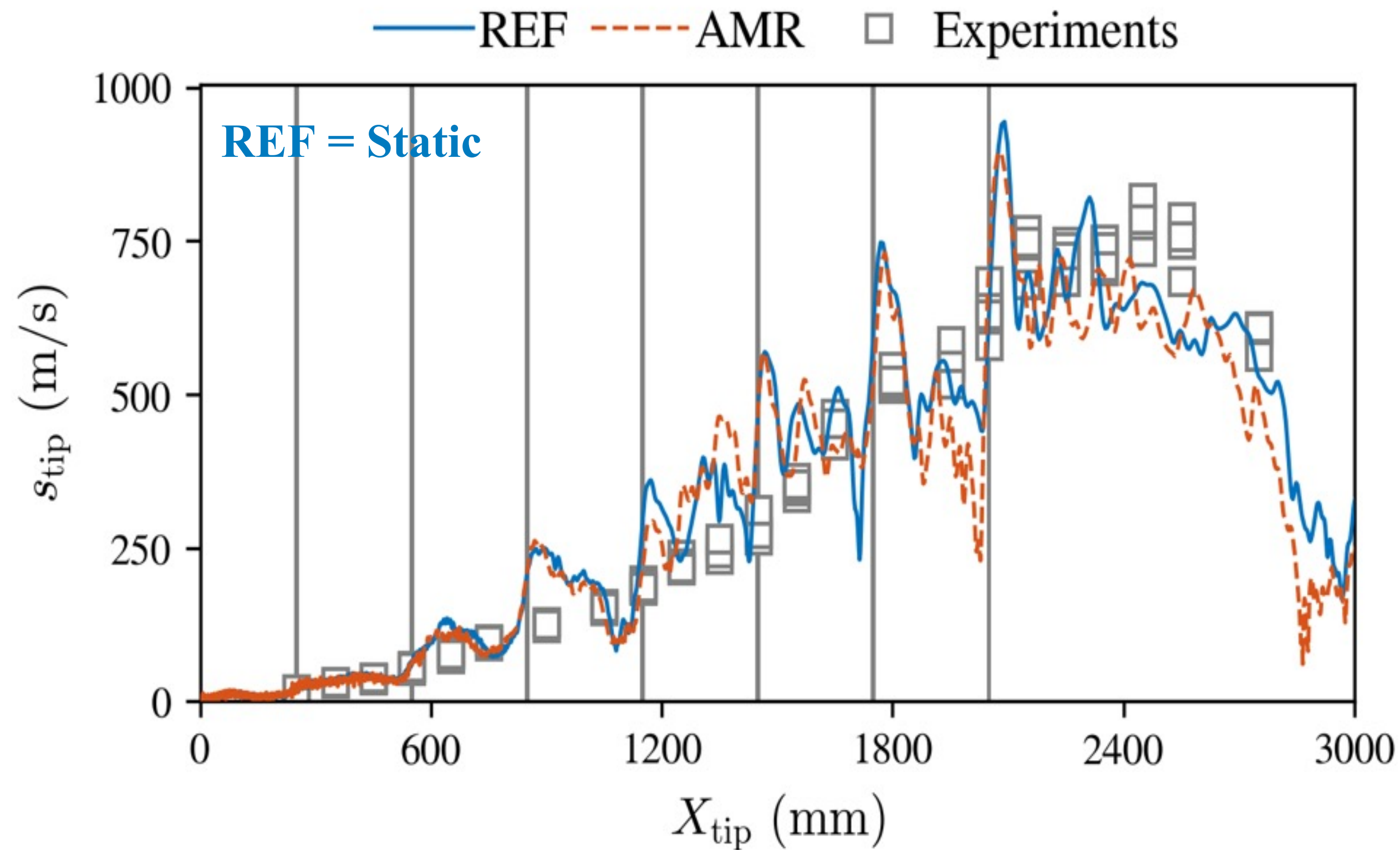
Validation of the AMR strategy



$t = 0.1 \text{ ms}$
 $X_{\text{tip}} = 0.01 \text{ m}$



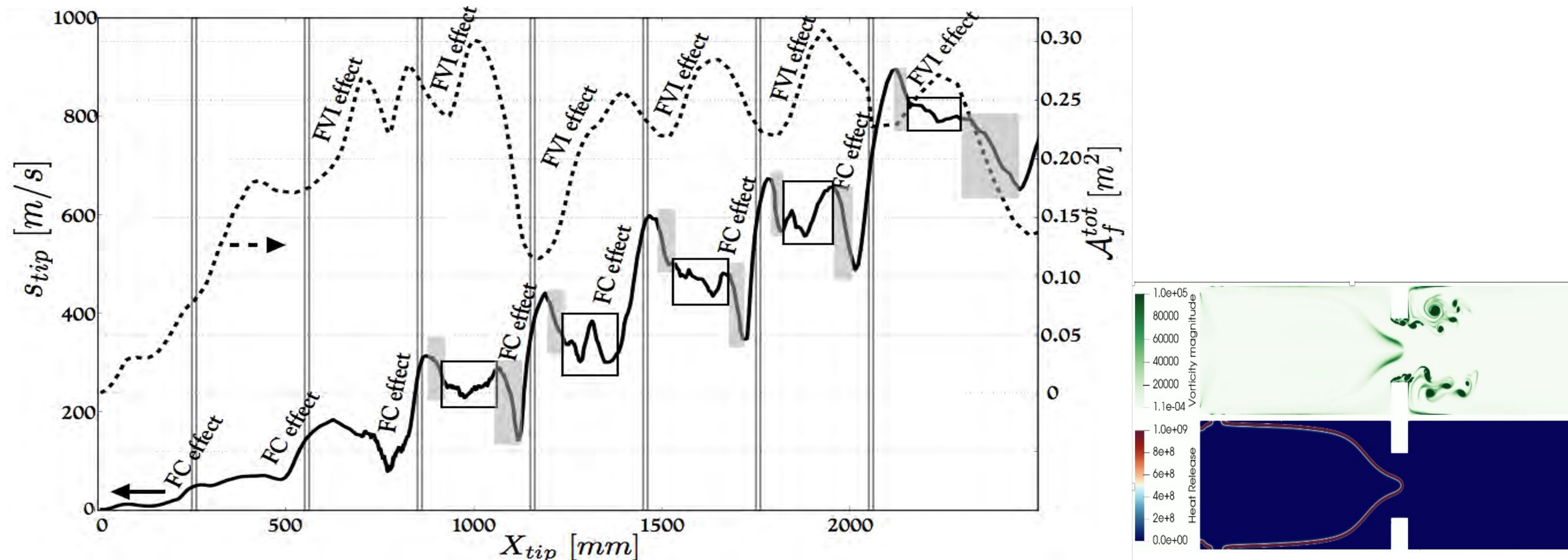
Validation of the AMR strategy



- Very good agreement between AMR and Static:
 - Flame acceleration
 - Shock formation and overpressure production

FA mechanisms

Flow contraction (FC) + Flame/Vortex interaction (FVI)

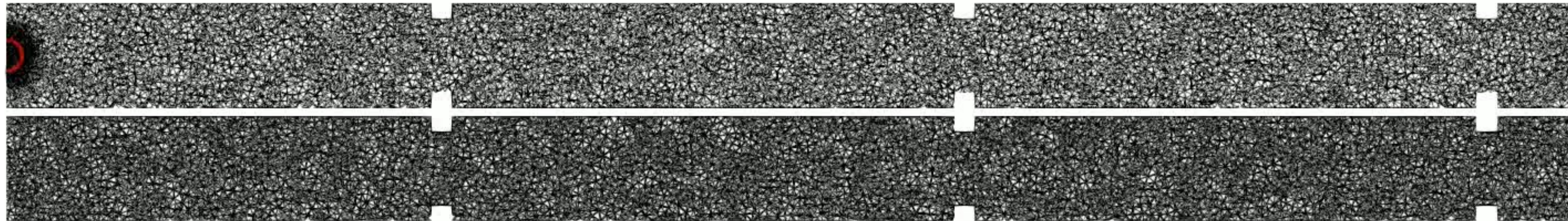


Flame propagation speed s_{tip} (solid line) and total flame surface A_f^{tot} (dashed line) as a function of the flame tip position X_{tip} .^[3]

[3] O. Dounia, Ph.D. thesis, INP Toulouse (2018).

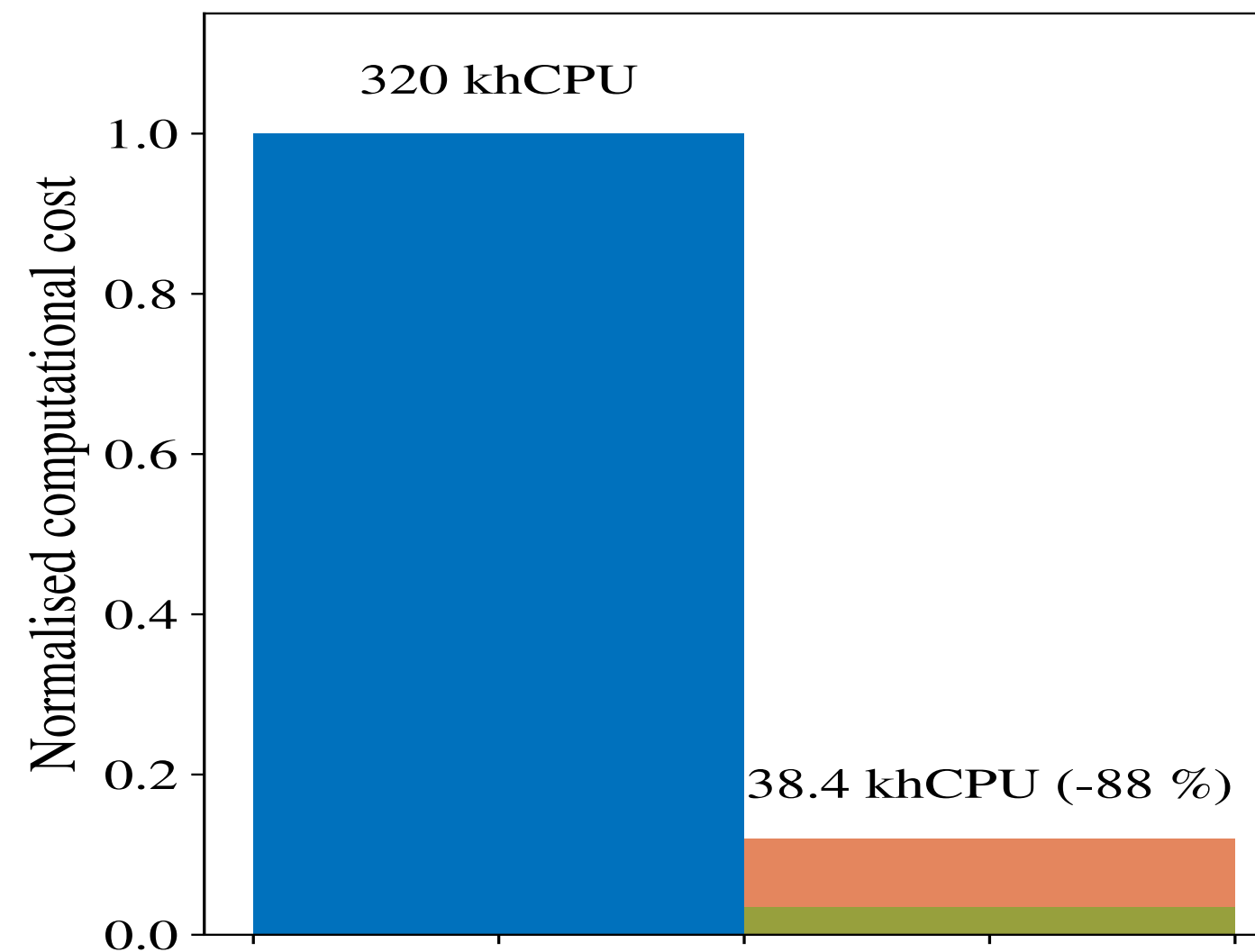
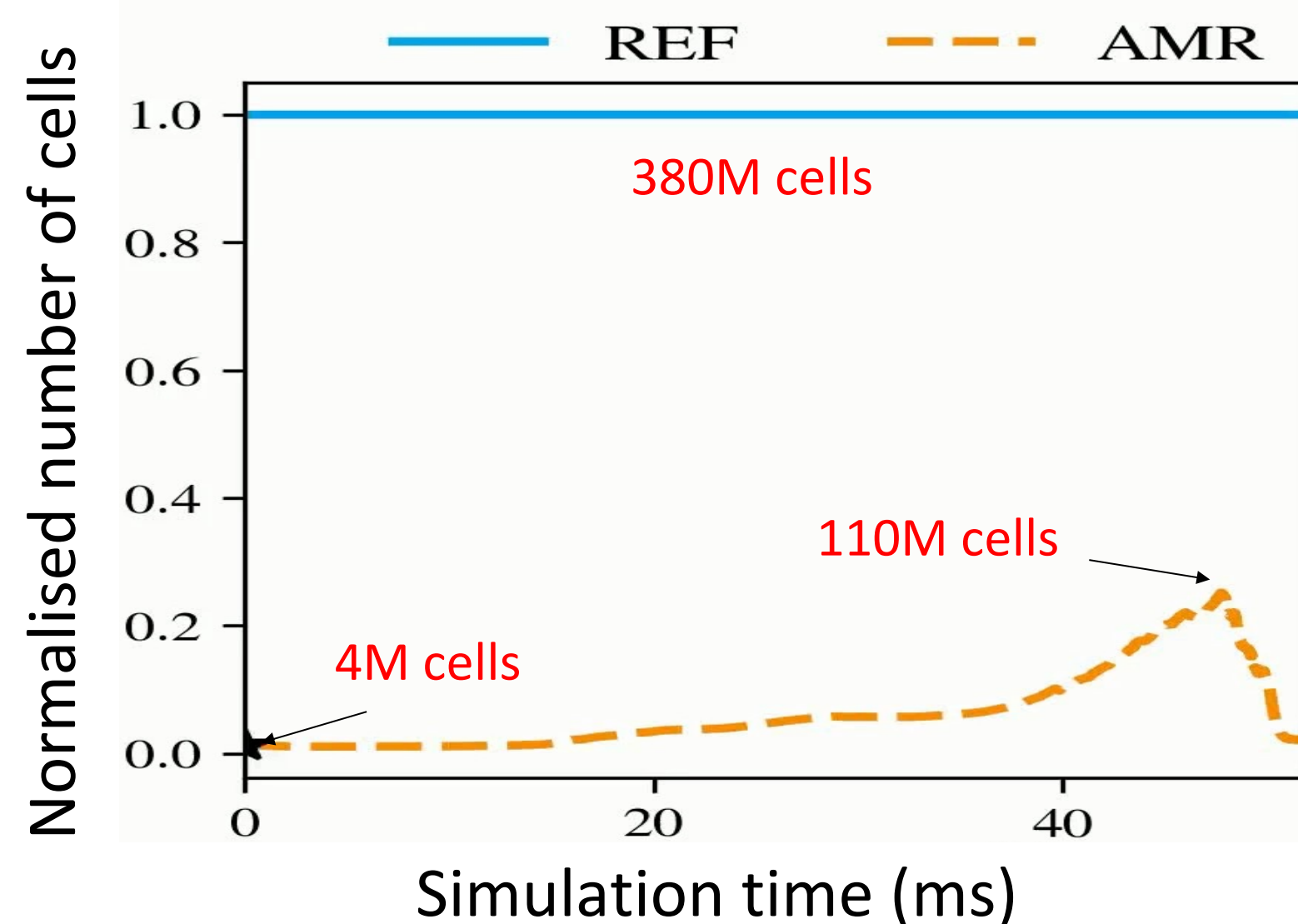
Computational efficiency of AMR

Time: 0.000100 s



- Increase of mesh size with FA
- **Mesh size $\leq 0.3 \times$ (Static mesh size)**

- 88% reduction in computational time
- **Cost of remeshing low**



- Static mesh solver cost
- AMR solver cost
- Mesh adaptation cost (1015 remesh)



Conclusion



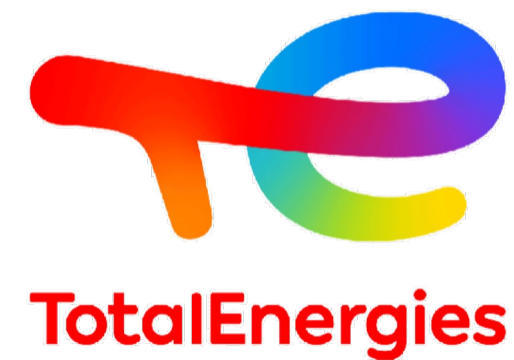
Takeaways

- A feature-based dynamic mesh refinement strategy was developed:
 - Suited for LES on unstructured grids
 - Does not rely on user-defined parameters for its sensors (flame, shocks, vortex structures)
 - Automatic remeshing triggering
- Validation against a fast deflagration case:
 - Good agreement with experiments and static mesh results
 - Drastic gain in computational efficiency compared to the static mesh approach
- Validation against other cases not shown here
- Working towards application in larger-scale explosions



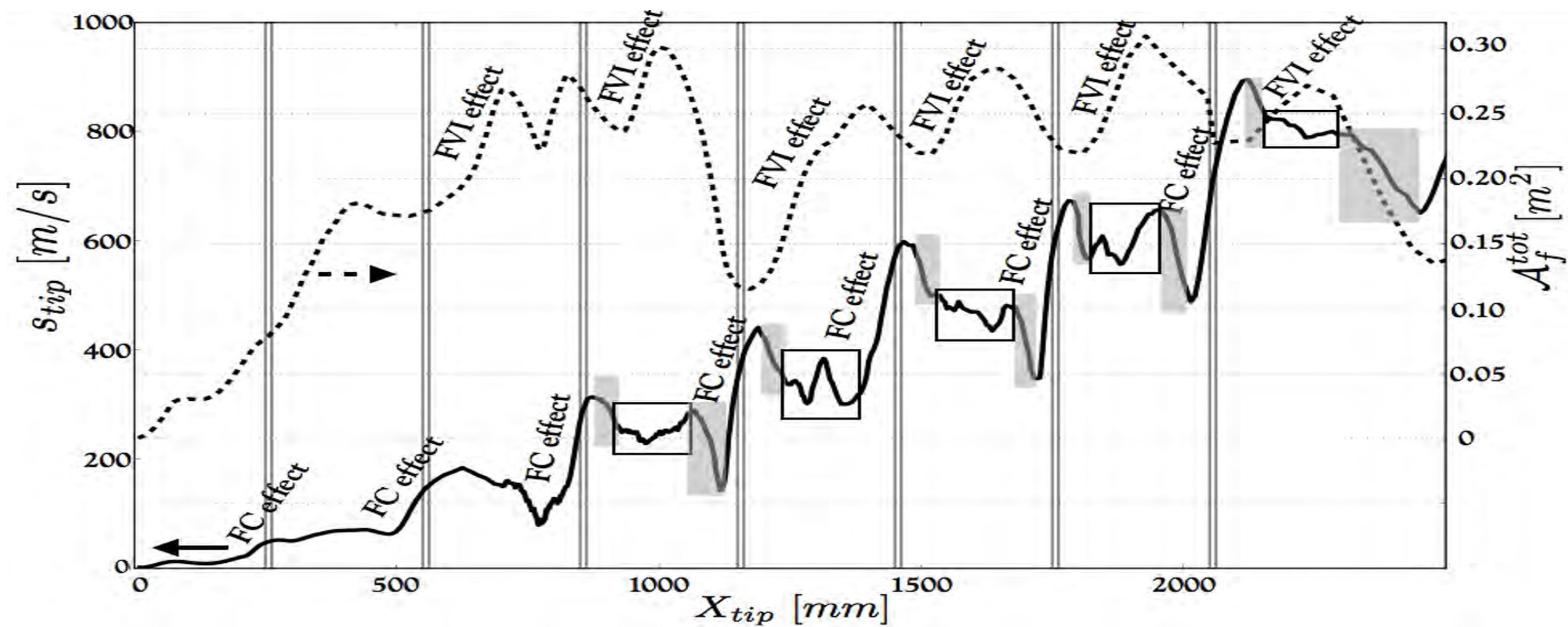
Funding acknowledgement

LES FOR EXPLOSIONS (LEFEX)

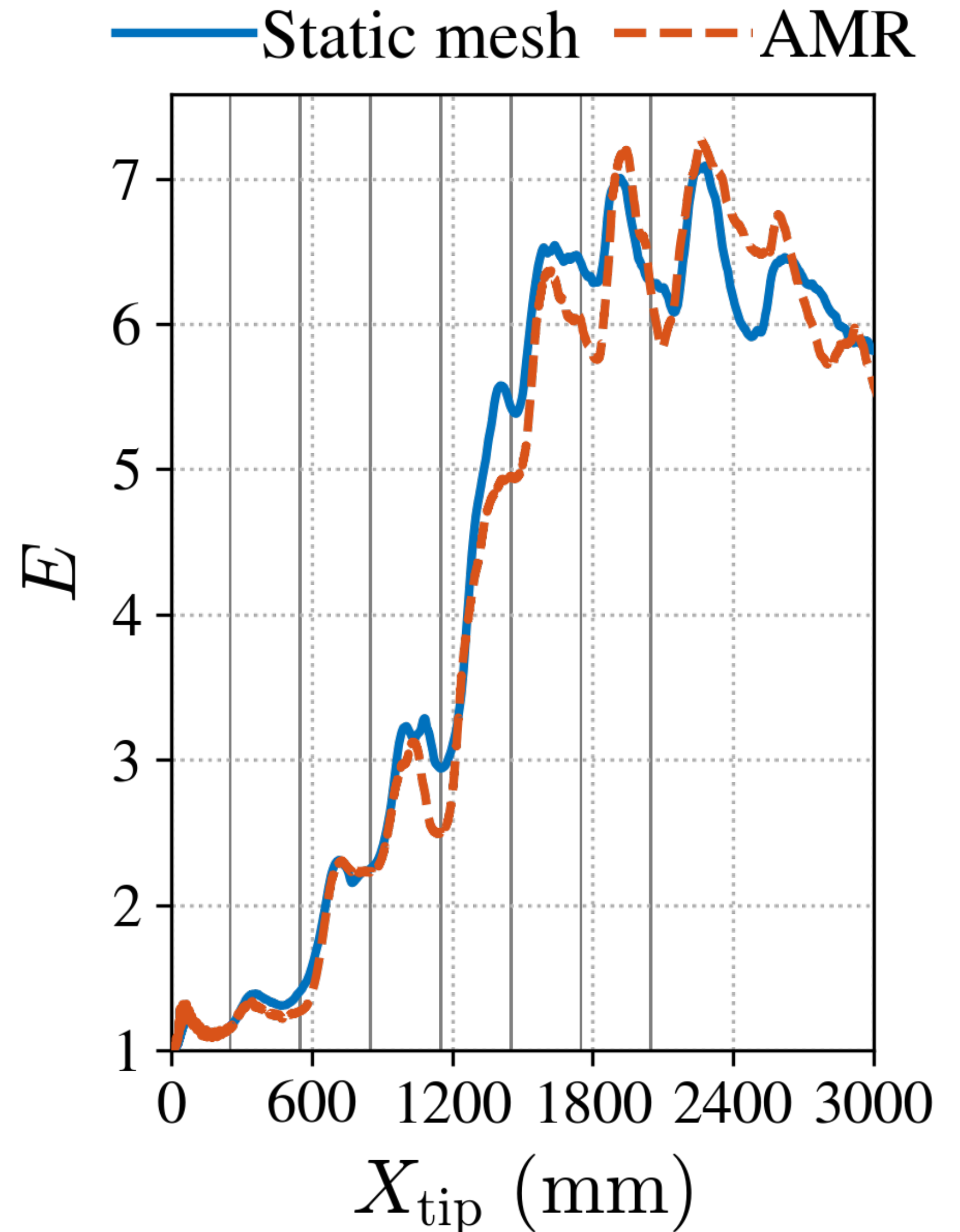


Goal of the project: Develop a high-fidelity LES code for explosion prediction in large/complex geometries.

Gravent BR30hS300 Test case^[2]



Flame propagation speed s_{tip} (solid line) and total flame surface A_f^{tot} (dashed line) as a function of the flame tip position X_{tip} .^[3]



[2] L. Boeck et al., The gravent ddt database, Shock Waves 26 (2016).
 [3] O. Dounia, Ph.D. thesis, INP Toulouse (2018).
 [4] L. Boeck, Ph.D. thesis, Technische Universität München (2015).

Parallel Kalpataru AMR library

Topology-Aware adaptive Refinement and load-balancing framework for Unstructured meshes

- Couples the load-balancing tool with the remeshing library MMG3D
- Iterative approach

